

Numerical study of the hydrodynamic interaction between ships in viscous and inviscid flow

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Abstract

A study was performed of the hydrodynamic interaction between a tug and a tanker ship model, using Computational Fluid Dynamics (CFD) code to calculate the hydrodynamic interaction force coefficients and the associated wave pattern generated by the two vessels.

The study was conducted for two velocities (full scale) of 4.0 and 6.0kn and depthdraught ratio 1.1 and 1.51, respectively. Two set of variables were considered, which are the longitudinal offset, and the lateral distance. A simulation of the Tug sailing freely was conducted to determine the pure interaction loads with or without deformable free surface.

Several CFD models were carried out using viscous and inviscid fluid flow, with and without taking into account the free surface. For viscous fluid flow the use of a zone with turbulent properties was simulated. Two turbulence models were used: The Shear Stress Transport, for computations with rigid free surface and the Standard Spartar Allarmar for the other case. The appropriate mesh size and time step were estimated based on previous sensitivity study of the mesh. Subsequently to obtaining these numerical results, the data were validated with experimental results.

The comparison between the numerical and experimental analysis showed in general good agreement. Two main aspects were found for the lateral distance variable by the CFD model evaluation. First, the free surface influence is much more important than viscosity, when the tug is close to the other ship, and secondly, an extremely fine was required mesh for the interaction ship zone.

Key words: Computational Fluid Dynamics, Hydrodynamics, Interaction between Ships, Free Surface Flow, Turbulence model

Resumo

Realizou-se um estudo da interacção hidrodinâmica entre um rebocador e um modelo de navio-tanque, utilizando Dinâmica dos Fluidos Computacional (CFD) para calcular os coeficientes das forças de interacção hidrodinâmica e o padrão das ondas geradas pelos dois navios. O estudo foi realizado para duas velocidades de 4.0 e 6.0Kn e a relação de profundidade e imersão de 1,1 e 1,51, respectivamente. Consideraram-se dois conjuntos de variáveis que são o deslocamento longitudinal e a distância lateral. Realizou-se uma simulação da navegação livre do rebocador para calcular as forças puras de interacção.

Realizaram-se vários modelos CFD utilizando escoamento viscoso e inviscido, com e sem superfície livre. Para o escoamento viscoso simulou-se a utilização de uma zona com propriedades turbulentas. Utilizaram-se dois modelos de turbulência: o "Shear Stress Transport", para os modelos sem superfície livre e o Standard Spartar Allarmar para o outro caso. A dimensão da malha e o passo de tempo foram estimado com base nos resultados de um estudo de sensibilidade da malha. Posteriormente os dados foram validados com resultados experimentais. A comparação entre a análise numérica e experimental mostrou uma boa concordância em geral. Da avaliação do modelo CFD obtiveram-se duas conclusões principais relacionadas com a variável distância lateral. Em primeiro lugar, a superfície livre é muito mais importante do que a viscosidade, quando o rebocador fica perto do outro navio, sendo necessária uma malha muito fina para a zona da interacção.

Palabra chave: Dinâmica dos Fluidos Computacional (CFD), Hidrodinâmica, Interação entre duas Embarcaçoes, Superficie Livre, Modelos de Turbulencia.

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Nomenclature

U	Velocity vector or speed of advance
$u_i = (u, v, w)$	Velocity components in the directions of $x_i = (x, y, z)$
р	Pressure
g	Gravity
τ	Stress tensor
$lpha_{_{q}}$	Volume Fraction
ρ	Density
ν	Kinematic viscosity
\boldsymbol{V}_t	Turbulent kinematic viscosity
k	Turbulent kinetic energy
ε	Turbulence eddy dissipation rate
ω	Dissipation rate per unit kinetic energy
$\widetilde{\mathcal{V}}$	Eddy-viscosity variable
_	
L_{pp}	Ship length
L _{pp} Re	Ship length Reynolds number
Re	Reynolds number
Re Fn	Reynolds number Froude number
Re Fn X	Reynolds number Froude number Surge force
Re Fn X Y	Reynolds number Froude number Surge force Sway force
Re Fn X Y N	Reynolds number Froude number Surge force Sway force Yaw moment
Re Fn <i>X Y N X</i> '	Reynolds number Froude number Surge force Sway force Yaw moment Surge coefficient
Re Fn X Y N X' Y'	Reynolds number Froude number Surge force Sway force Yaw moment Surge coefficient Sway coefficient
Re Fn X Y N X' Y' N'	Reynolds number Froude number Surge force Sway force Yaw moment Surge coefficient Sway coefficient Yaw coefficient
Re Fn X Y N X' Y' N' ξ	Reynolds number Froude number Surge force Sway force Yaw moment Surge coefficient Sway coefficient Yaw coefficient Longitudinal distance
Re Fn X Y N X' Y' N' ξ η	Reynolds number Froude number Surge force Sway force Yaw moment Surge coefficient Sway coefficient Yaw coefficient Longitudinal distance

CHAPTER 1 Introduction

1.1 General

Many problems that interfere with navigation are due to the manoeuvres of ships, and one of these problems is the hydrodynamic interaction between ships, which occur frequently due to increase in the ship traffic density. According to UNCATD (2003) the maritime transportation represents more than 90% of the world fleet, and the world fleet in 2002 becomes more than 800 million Dwt. In 2008 the merchant vessels higher than 100 GT were 99.741 ships completing 830.7 million GT between them (Fairplay World Fleet Statistics, 2008).

The interaction problem in navigation is usually produced when the ships are moving in restricted waterways, such as harbours or canals (King, 1977). The encounter of two vessels can fall into one of two main categories. In the first case, a ship passing another one at a close distance, which commonly happens when sailing in narrow channels. In the second case, a ship manoeuvring very close to another one due some routine operation, like the tug assistant. In the interaction problem the flow around the ship hulls is modified, generating additional forces in the horizontal plane on the ships (surge and sway forces, and yaw moment). Some examples are show in Fig 1.1.

The interaction phenomenon is also influenced and caused by the two navigation boundaries which are: the bottom, and the lateral boundaries of the navigation area. The former, is usually given by introducing depth dependent hydrodynamic coefficients. The latter is limited by bank or quay walls, causing the so-called bank effect to a ship navigating in parallel course thus, producing the hydrodynamic interaction forces on a ship in a channel towards or away from the nearby obstacles (Ch'ng, 1991, 1993). Some examples can be seen in Fig 1.1.

In restricted water, the shallow water condition has a strong influence in the hydrodynamic interaction forces (Fortson, 1969). In shallow water, the average depth is about 10 to 20 m, which is usually smaller that the ship length (50 to 100 m). Here, the wave generated by the ship has a larger length than the wave generated in deep waters at the same velocity. Thus, the magnitude of the wave length in general may be considered similar to the ship length and also the wave can show an anomalous wave height. The generated ship waves are hardly dissipated when interacting with the shorelines, affecting the water conditions of the navigation. In case of interaction, the effect on each vessel increase, either in a moored or the sailing vessel.

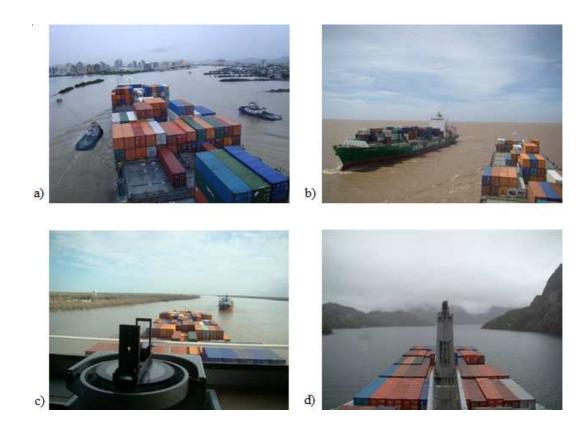


Figure 1.1 Interaction between ships and their boundary: a) A vessel is assisted by a tug near the harbour; b) Two ships sailing in a river in head encounter; c) Manoeuvring of overtaking between two ships in calm water; c) Ship sailing in a narrow canal.

The phenomenon produced in the interaction problem can cause serious accidents, when it is not considered. Chatterton (1994) comments the famous accident of Queen Elizabeth II. Chatterton describes that the vessel was sailing at high speed in shallow waters and then the suction force between the bottom and the ship caused the Queen Elizabeth II to run aground off the Cutty Hunk Island.

The Marine Accident Investigation Branch (MAIB) reports the maritime accidents yearly. In their reports the accidents due to the interaction between ships are the most common ones. A collision reported by MAIB, between MV Asch and MV Dutch Aquamarine in the South West lane of the Dover Strait TSS, with the loss of one life and which, occurred in October of 2001. They found that the cause of the collision was because the two vessels were on coincident tracks and travelling at different speeds. The interaction between the ships was described as follow: "The two vessels became very close it was apparent from witness observations that Dutch Aquamarine's track was, in fact, a few meters to starboard of Ash's. As Dutch Aquamarine's bow approached Ash's stern on her starboard quarter, hydrodynamic interaction caused Ash's heading to alter to starboard. The flare on the port side of Dutch

Aquamarine's bow first made contact with the extreme starboard quarter of Ash's bridge deck, causing damage to railings, the lifeboat and its davit arm."

When the differences in the ship dimensions are large, the effect produced during the interaction between ships increases and the risk of accident is higher for the smaller ship. A typical situation which involves differences in ship dimensions is the ship-tug assistance. When a tug assists a ship, the position of the tug with respect to the assisted ship and the lateral distance, can be constantly changing. The consecutive positions of a tug when is approaching to assist a ship, are shown in Fig. 1.2.

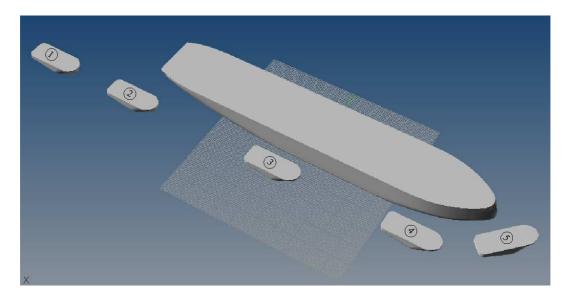


Figure 1.2 Different position of the Tug assisting the merchant ship in a harbour

When the tug is near the stern of the ship (position 1), an increase in its velocity may occur due to the flow velocity from the aft of the ship. In close proximity to the ship hull, a low pressure starts moving the tug in the ship's direction. For ships in ballast condition, or ships having particular overhanging stern, the tug can easily go to the position 2, generating damages to its hull or superstructure.

Going forward and near the hull (position 3), the tug is in under action of an important suction force in the direction of the ship hull, and negative yaw moment (according with right-handed Cartesian frame of reference) is due to the accumulated water in the tug bow. When the tug is attracted by the ship, it is in general difficult to recover its course. When the tug is in position 4 (side of the bow) she enters on area of high-pressure the negative yaw moment is growing, and must be compensated by the appropriate use of the rudder and propeller to avoid the risk of accident.

In position 5 when the tug is near the bow, a strong a negative sway force acting on the stern brings the tug to the front and under the bow with the risk of capsizing. Then, proper operational condition must be applied

The study of the interference when a tug is operating near to another ship is important to define the prediction of the manoeuvring characteristics of the tug and is useful to optimize the waterway operation. Therefore, developing a model able to predict the interaction forces with accuracy and considering restricted waters, and course keeping of ships is necessary. The hydrodynamic interaction prediction of a ship in restricted waters has been investigated for several decades, as is described in the following section. Trough, it is necessary to perform more detailed studies.

1.2 Objectives

The main objective of this work is to develop numerical simulations of manoeuvring operations of a tug interacting with a ship in shallow water. The Computational Fluid Dynamics (CFD) software is used for the computation, which is able to predict the hydrodynamic interaction forces, moment, and the behaviour of the free surface, in the general situation of interacting ships.

This thesis presents general observations on the tested models considering the hydrodynamic interaction between two ships in movement. The hull forms tested are a typical tanker ship and a typical tug vessel. Special emphasis is set in the range of the side distance between ships, and the relative heading angle of the supply vessel.

The ship manoeuvres that were selected to be simulated in the present study are the simulation of the flow around both ship hulls in a parallel course sailing at the same velocity. Two cases of tug position along the length of the tanker and two cases of velocity are studied. Several variations of side distances are considered to obtain the favourable case considering the interaction forces.

The obtained results are useful to determine the characteristics of the interaction phenomenon, and to obtain a practical solution on interaction between ships, providing data for validation of numerical fluid dynamic simulation.

Another aim of this dissertation is to show the potential of CFD to solve manoeuvring problems in naval architecture and marine engineering with accuracy and short resources and time. Thus, the CFD code can be presented as an efficient tool in the ship hull design.

1.3 Organization of the thesis

The present thesis consists of four main chapters:

Chapter 1 *Introduction*: Here it is presented the motive, objectives, and general presentation of the interaction problem, providing the state of the art of the hydrodynamics interaction forces necessary for carrying out.

Chapter 2 *Problem Statement*: This chapter describes the theoretical background and a review of the method used to solve the hydrodynamic interaction problem, explaining the basic principles of the governing equation and the numerical method to solve.

Chapter 3 *Application of the numerical method*: The problems under study are presented. Defined are all parameters used in the Computational Fluid Dynamics simulation, including general criteria, boundary conditions, the configuration of the mesh and the volume domain.

Chapter 4 Analysis of the numerical results: Finally, the results are summarized and discussed giving the most important conclusions of this study and proposing further improved work.

1.4 State of the art: Ships hydrodynamics interaction

The phenomenon of hydrodynamic interaction between ships is a subject of several research works. The interest to study the interaction problem started around the 1900's, considering primarily experimental efforts dealing with the interaction force and moments developed between ships in proximity. One of the first experiments performed and reported on the interaction between ships was conducted by Davis W. Taylor (1909), who explained the suction that tends to bring ships together when they are passing close to one another. His investigation was focused on the quantitative measurement, and was carried out for two models. The first test was with a relativity narrow models, with the following dimensions for each ship: Ship (A): length = 20.512m, breath = 3.69m, draft = 1.26m; Ship (B): length = 20.512m, breath = 3.50m, draft = 1.20m. The second test was with fuller models, whose dimensions were: Ship (A): length = 20.512m, breath = 3.59m, draft = 0.96m; Ship (B): length = 20.512m, breath = 2.512m, breath = 3.59m, draft = 0.96m; Ship (B): length = 20.512m, breath = 2.78m, draft = 1.24m. Taylor stated that the models were tested at speeds between 2.0 and 3.0Kn. The total forces on the models were determined by adding the measured forces at each end of model. He also comments that the results may have

inaccuracies due to problems encountered in the models on a parallel course and stated that the results for the thinner models were less consistent.

After the test of Taylor, several experimental works have been carried out with more accuracy, describing the hydrodynamic interaction forces. Newton (1960) based on the studies of Taylor, investigated the interaction effects during overtaking manoeuvres by experiments between two ship models in deep water, The tests conducted include an experiment with scaled models of the battleship H.M.S. KING GEORGE and the R.F.A. OLNA. (Ship (A) and Ship (B) respectively) the characteristic of these ship were: ship (A): length = 4.51m, breath = 0.63m, draft = 0.18m, $C_B = 0.61$; ship (B): length = 3.46m, breath = 0.43m, draft = 0.18m, $C_B = 0.71$. The experimental tests were carried out using freely propelled and controllable models to study the behaviour during manoeuvring alongside and breaking away. Newton also conducted experiment at sea with real ships, giving valuable data in this respect.

Later the experimental tests were performed to study the influence of the restricted water on hydrodynamic interaction forces Müller (1967) studied the overtaking and encounter of ships in a narrow canal. New variables were considered by Remrey (1974) investigating the effect of the size passing vessel and the separation between the passing and moored vessel.

In parallel with the experimental tests, two numerical methods based in the slenderbody and the potential theory were developed, to calculate the interaction sway forces and yaw moment. These methods have been used extensively for many authors giving important information, to understand the behaviour of the hydrodynamics interaction between ships.

Newman (1965) used the slender-body theory to develop a code that can solve the hydrodynamics interaction problem. Newman supported his work assuming that the fluid flow is ideal, incompressible, and unbounded except by the wall and the body. He also assumed that the body is axisymmetric and slender if r/L << 1, and the body is in proximity to the wall if z/L <<1, (here *r* is the radius of the body, *L* is the body length, and *z* is the distance from the centerline of the body to the obstacle). The aerodynamic theory used the source distribution to calculate inside the body. For allowing the presence of the wall, the source distribution was offset from the body axis. The body was considered to be fixed in space, with a free stream flowing past in the opposite direction to the body. Legally's theorem (1953) was used to calculate the force and moment. The results of Newman have been extensively used for evaluating the result of other methods.

Tuck (1966) extended the slender-body method to study the problem of the hydrodynamic interaction in shallow water flow passing a fixed slender obstacle in a stream. He provided a model for the behaviour of ships moving in still water with restrictions in depth, which was subsequently applied to a variety of problems involving shallow water, such as river flow past obstacles. Tuck (1967) also extended his earlier method and introduced the concept of effective width in order to deal with ship dynamics in shallow water of restricted width.

Yeung (1978) used the shallow-water sources, including the effect of circulation, to calculate sway forces and yaw moments on each ship. Davis and Geer (1982) developed an alternative method for calculating the slender-body sway forces and yaw moments, based on an asymptotic analysis.

King (1977) developed a technique for the unsteady hydrodynamics interaction between two ships moving along parallel paths in shallow water using the slender-body theory. The problem was analogous to that for two porous airfoils passing each other. A system of singular integral equations was derived and solved numerically. The sway force and yaw moment were calculated for each ship. For testing this model and numerical procedure, a comparisons was made with publishes experimental and theoretical results for the far field obtaining good agreement.

In the other hand, Fortson (1969) developed a method to calculate the sway forces and yaw moment, assuming that, when ships are close together, the potential field is the main source of the interaction force. The flow around the ships was represented using half bodies of revolution generated with axial singularity distributions. Axial distributions of doublets were included to correct for cross flow and thus maintain the rigid body boundary condition. The main result of Fortson was a computerized model to predict forces resulting from steady parallel motion of bodies of revolution moving in an infinite ideal fluid. The code was capable of handling ship geometries in rectilinear motions. It was found that the theoretical model produced similar results when compared with model test conducted by Taylor (1909). Fortson concluded that is necessary to obtain a set of more valid empirical data.

Ashe (1975) continued the work of Fortson assuming that interaction forces arise only from disturbance and therefore, a rigid free surface can exist, and the fluid flow is ideal and infinite therefore, potential theory is applicable. As well as with Fortson (1969), Ashe used an axial distribution of dipoles to take into account the wall effects. As in the previous methods,

Ashe also uses Legally's theorem to calculate the forces and moments on the body, but he includes the unsteady terms. Here the steady-state force is due to interaction effects as shown by D'Alembert paradox. Ashe extended the theory presented by Fortson and included the unsteady effects associated with the acceleration, vertical velocities, and rotational velocities in a quasi-steady state.

Beck et al. (1975) extended the analysis of interaction between ships including the case of a ship operating in a dredged channel surrounded on both sides by shallow water. Beck (1977) examined the case of a ship travelling parallel to, but displaced from, the centreline of a shallow channel.

Chatterton (1994) presented an evaluation of the proximity codes. These codes were developed to predict the interaction force moment experienced by an Unmanned Underwater Vehicle when is in proximity with an obstacle. The theory used a potential flow model of a body operating in an ideal fluid based in the Ashe (1975) theory. The codes were extended including unsteady motion, inclined bottom, and automatic convergence. The validation of these codes involved performing model testing at the Massachusetts Institute of Technology (MIT) towing tank as well as collecting previously published data examining the theoretical trends. A theoretical database was compiled to examine the trends associated with changes in vehicles shape and orientation.

In order to improve the accuracy of the prediction of interaction forces and moments between ships, a semi-empirical method was developed by Brix (1993), estimating the time histories of the forces and moments in the horizontal plane due to interaction with another ship as a function of geometry speed and environmental parameters based on his previous work (Brix, 1979). The method to estimate the forces and moments acting on a ship during an overtaking manoeuvre was presented in the Manoeuvring Technical Manual. Here approximations were formulated for the maximum values of the longitudinal and transverse forces and for the yawing moment. The method was subjected to some restrictions: it was only valid for overtaking manoeuvres. Thus, the influence of water depth was not taken into account, and also the ratio of ships' lengths is limited. Brix states: "Besides some theoretical approaches and experimental results no reliable results are available except of a semi-empirical nature."

Varyani et al. (1999) presented empirical formulae for predicting the peaks of the lateral force and the yawing moment during interactions between two ships. Here the validity

of the method was also restricted being: the cases $V_T = 0$ or $V_O = 0$ were not covered, the length ratio was limited, and the method was only valid for encountering manoeuvres.

Vantorre et al. (2002) reported results from a large series of ship-ship interaction model experiments using an empirical method to calculate the extreme peaks in typical time traces of interaction forces. The investigation was carried out for four ship models in shallow water towing tank, covering a large variety of parameters such as overtaking/overtaken, speeds, distances, and water depths. In the work they suggested that it is an impossible develop a full empirical method taking into account all the possible parameters that influence the interaction forces between two ships passing each other.

The mooring forces on a container ship induced by a passing bulk carrier were reported by Varyani et al. (2003). The mooring system was modelled as linear and the results were obtained using an empirical method for calculation of ship-ship interaction forces acting on a ship at zero speed. Similar works have been presented by Krishnankutty and Varyani (2003, 2004).

Recently, new experimental tests have provided fresh information, about of the parameters and variables that influence the interaction phenomenon. Kyulevcheliev et al. (2003) presented a general overview and initial findings of a set of model experiments on hydrodynamic impact of a moving ship on a stationary one in restricted water. They found that the wave effects increase drastically the peak values of the interaction loads and change substantially their time histories.

Taggart et al. (2003) examined experimentally the hydrodynamic interactions between ships that can influence seakeeping during operations. In the experiments a semi captive model was used, and the numerical code was modified to include restraining forces for specific modes. The numerical predictions and the experiments showed that the presence of a larger ship can significantly influence the motions of the smaller ship in close proximity.

Kriebel (2005) carried out an experimental measurement of the loads on a moored ship resulting from a passing ship moving in parallel to the moored vessel. Variations in the model tests were included which consider changes in the passing vessel speed, vessel displacement, water depth, and separation between the two ships. The experimental data were analyzed in two manners. First, the empirical equations were developed, describing variation in the peak mooring loads with changes in the parameters. Second, two existing models were evaluated in blind test to determine their ability to predict the measured mooring loads. With the new information and mathematical techniques, the numerical solvers have been improved, expanding their application and allowing performance of other situation, such as seakeeping during the interaction phenomenon. Most of these codes follow the trend of the modern hydrodynamic research works, which are based in the potential theory and solved by the panel method. Chen and Fang (2001) used a 3D panel method based on the source distribution technique to analyze the hydrodynamic interaction problem between two moving ship in waves. The numerical solution was evaluated by applying the method to two pairs of models and compared with experimental data and the strip theory. From the comparisons the result showed that the hydrodynamics interaction calculated by 3D method is more reasonable. However, it is not so significant using the 2D method.

Varyani et al. (2002) developed a potential theory method for calculation of ship-ship interaction forces. The method was validated against results of Dand (1981) and Yasukawa (1990). The results were presented for two and three, ships meeting in channel.

The three-dimensional (3D) panel method based in the potential theory was developed by Yasukawa (2003) to calculate the ship-ship interaction forces between two ships in an overtaking situation. The motions were simulated using coupled equations of motion and the 3D panel method provided hydrodynamics interaction forces and added masses as function of the ships relative positions.

Sutulo and Guedes Soares (2009) created a module of a manoeuvring simulating system using potential forces based on the Hess and Smith panel method aiming to calculating the hydrodynamics interaction forces. The comparative computations of the added masses, surge and sway interaction forces and yaw interaction moments with varying number of surface computational panels showed that on a typical modern computer, an acceptable accuracy in terms of the integrated loads can be reached with a relatively small number of panels allowing real-time simulations with the developed algorithm in the loop. Comparisons with available experimental data and with an empirical interaction prediction method demonstrated that the code works adequately and can be considered as validated, although disagreements with the experimental data were observed in the cases when the viscous effects were significant.

Other methods have been applied in the interaction phenomenon, obtaining good result, e.g. the Wave Equation Model (WEM) developed by Jiankang et al. (2001), which solve the shallow water equations with moving surface pressure. The model was applied to a

numerical investigation of two moving ships Series 60 with a $C_B = 0.6$. The ships have streamline-stern or blunt-stern moving in parallel. The 3D wave interaction patterns were investigated, finding that the wave interaction occurs mainly behind the ships.

A higher-order boundary element method (HOBEM) combined with generalized mode approach was applied by Hong et al. (2005) to analysed of motion and drift forces of side-byside moored multiple vessels (LNG FPSO, LNGC and shuttle tankers). The models were compared with experimental result finding good agreement between the methods.

The oldest method of Newman has been updated for applications in new ships interaction situation. Gourlay (2009) investigated the sinkage and trim of two moving ships when they pass each other, either from opposite directions, or when one ship overtaking the other. The work was simplified to open water of shallow constant depth. The investigation was carried out using linear superposition of slender-body shallow-water flow solutions. It was shown that even for head-on encounters, oscillatory heave and pitch effects are small. Also it was found that the sinkage and trim can be calculated using hydrostatic balancing. The results were compared with available experimental data, on a container ship and a bulk carrier in a head-on or overtaking encounter.

The recent advances in hydrodynamic theories and computation capacity have been developed of high level Computational Fluid Dynamics (CFD) simulators for design applications, which allow the use of this tool to solve some manoeuvring problems. Wnek et al. (2009, 2010) presented the results of a CFD analysis of the wind forces acting on a LNG ship and a barge model. The comparisons between the numerical and experimental result were in good agreement.

Burg and Marcum (2003) developed a nonlinear scheme based on Farhat's algorithm to calculate the free surface elevation for unsteady Reynolds-Averaged Navier-Stokes (RANS) code. The results for a DTMB Model 5415 in small drift angles, constant turning radius and a prescribed manoeuvre although fairly good, were not mesh independent and showed that more work is needed on the turbulence models and grid deformation rate.

Lee, et al (2003) used both experimental and unsteady RANS approach to obtain quantitatively the hydrodynamics forces under the lateral motion, and a view to generalize the obtained hydrodynamics forces that in a practical use. The water depth was an important factor to exercise the influence on the inertial forces and transitional lateral forces acting on ship hull. They concluded that using a concept of circulation is effective to express the lateral drag coefficient qualitatively.

Some authors have been used the CFD codes to solve the problem of interaction between ships. Chen et al. (2003) present ship-ship interaction forces calculated from an unsteady RANS code comparing the ship-ship interaction model test results performed by Dand (1981). Good agreement was found between the calculations and the measurements. They concluded that the free surface is important for the interaction between the two ships.

Huang and Chen (2003) presented potential hazards to vessels moored at nearby piers. A Chimera RANS code was used to explore the fluid activities induced by passing ships and their impacts to moored vessels at a complex waterfront next to a navigation channel. This model captured details of ship-ship and ship-pier interactions evolving the time, and taking into full account the viscous flow physics, exact seafloor bathymetry and basin boundaries, as well as nonlinear couplings between moored ship and pier. The results of the simulation clearly show that the significance of site specific factors to the dynamics of moored vessels induced by passing ships.

CHAPTER 2 Problem statement

In the previous chapter, the interaction phenomenon was presented, showing typical conditions in which the encounter of ships can occur. The effect of interaction between ships is based on reviewing previous research work, presenting a series of methods that can be used to solve the interaction problem, accurately. It is mentioned that a numerical solution to the Navier-Stokes equation can be applied to solve many manoeuvring problems. Thus, the hydrodynamic interaction between ships can be solved by a Computational Fluid Dynamics (CFD) code, even though a considerable computational cost is involved.

In the present chapter, the theoretical background of CFD for solving threedimensional (3D) fluid flow problems is introduced. The CFD code is extended from different solver techniques, discussing influences and impacts on the fluid flow simulation.

The chapter is divided in three main sections. The first one is an outline of mathematical equations and the boundary conditions, describing the fluid flow. The second section discusses various turbulence models, with special emphasis to the mathematical formulation of the Reynolds-Averaged Navier–Stokes (RANS) method. The third section is a review of the numerical methods and solvers describing the characteristic of CFD in the fluid flow study; the section also summarizes the theoretical background for Finite Volume (FV) method.

2.1 Governing equations

The continuity equation for the mass conservation and the Navier–Stokes equations, for the momentum transport, govern the 3D motion of the incompressible and viscous fluid. These equations are expressed in terms the set of partial differential equation as follows:

$$\frac{\partial u_i}{\partial x_i} = 0 \tag{2.1}$$

$$\frac{\partial(\rho u_i)}{\partial t} + \frac{\partial(\rho u_j u_i)}{\partial x_j} = \frac{\partial \tau_{ij}}{\partial x_j} - \frac{\partial p}{\partial x_i} + \rho g_i$$
(2.2)

where ρ is the density, p is the pressure, u_i is the velocity in the stream direction, τ_{ij} is stress tensor, and g is the component of the gravitational acceleration g in the direction of the Cartesian coordinate x_i . In the case of constant density and gravity, the term ρg can be written as grad $(\rho g \cdot r)$, where r is the position vector, $r = x_i$, i_i (usually, gravity is assumed to act in the negative z-direction, i.e. $g = gz_k$, g_z being negative; in this case $g \cdot r = gz_z$). Then $-\rho gz_z$ is the hydrostatic pressure, and it is convenient for numerical solution more efficient to define $\tilde{p} = p - \rho gz_z$ as the head and use it in place of the pressure. The term ρg_i then disappears from the above equation. If the actual pressure is needed, one has only to add ρgz to \tilde{p} .

The homogenous multiphase Eulerian fluid approach utilizes the Volume of Fluid method (VOF) to describe the free surface flow problem mathematically. The VOF method developed by Hirt and Nichols (1981) is a fixed mesh technique designed for two or more fluids, where in each cell of a mesh it is necessary to use only one value for each dependent variable defining the fluid state. The variables can be defined by a function α_q (where q represents the fraction of volume; q = 1, 2...). The value of this variable is one at any point occupied by the fluid and is zero in other cases and is writing as:

$$\sum_{q=1}^{n} \alpha_q = 1 \tag{2.3}$$

the average value of α in a cell represents the position of the interface of the fluid. In particular, a unity value of α corresponds to a cell full of fluid, while a zero value indicates that the cell is empty, so the cells with α values between zero and one must contain a free surface. The tracking of the interface is obtained by solving the continuity equation of the volume fraction. For the *q* liquid, troughs of the density as follow:

$$\rho = \sum \rho_q \alpha_q \tag{2.4}$$

For a Newtonian fluid the viscous stress tensor can be expressed as:

$$\tau_{ij} = v \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} - \frac{2}{3} \delta_{ij} \frac{\partial u_k}{\partial x_k} \right) = 2v \left(s_{ij} - \frac{1}{3} \delta_{ij} \frac{\partial u_k}{\partial u_k} \right)$$
(2.5)

where s_{ij} is the strain-rate tensor and ν is the dynamic viscosity. For an incompressible fluid, Equation 2.5 reduces to:

$$\tau_{ij} = 2\nu s_{ij} \tag{2.6}$$

the Navier-Stokes equations are a highly non-linear system. The strong non-linearity of the equations produces high frequency oscillations when the Reynolds number is increased, and the flow becomes unstable and turbulent. It is computationally very expensive to solve the

equations directly, which makes that presently, only in very simple geometry configurations it is possible to solve the Navier-Stokes equations using direct methods (DNS). The most common approach at the moment in hydraulic engineering practice is to solve the Reynolds Averaged Navier-Stokes.

2.1.1 Reynolds Averaged Navier-Stokes equations

The Reynolds Averaged Navier-Stokes (RANS) equations have been developed based on the concept that a velocity and a length scale are sufficient to describe the effect of turbulence in a flow. For instance, eddy viscosity model, estimates the velocity and length scales of the flow from the local mean flow quantities, which is done by relating the turbulent viscosity to the mean velocity gradient of the flow. However, the RANS model may fail to simulate more complex flow (Liaw, 2005).

In a statistically steady flow, every variable can be written as the sum of a timeaveraged value (\overline{U}) and a fluctuation (u') about that value:

$$U = \overline{U} + u' \tag{2.7}$$

where the time-averaging velocity component is defined as:

$$\overline{U} = \frac{1}{T} \int_{0}^{T} U dt$$
(2.8)

where T is the averaging interval. This interval must be large compared to the typical time scale of the fluctuations. Thus, if T is large enough, (\overline{U}) does not depend on the time at which the averaging is started.

If the flow is unsteady, time averaging cannot be used and it must be replaced by ensemble averaging described by Ferziger (2002) defining as follow:

$$\overline{U} = \frac{1}{N} \sum_{n=1}^{N} \overline{U}$$
(2.9)

where N is the number of members of the ensemble and must be large enough to eliminate the effects of the fluctuations. This type of averaging can be applied to any flow. The term Reynolds averaging is used to refer to any of these averaging processes; applying it to the Navier-Stokes equations yields the Reynolds-averaged Navier-Stokes (RANS) equations.

Averaged continuity and the momentum equations can, for incompressible flows without body forces, be written in tensor notation and Cartesian coordinates as:

$$\frac{\partial \overline{u_i}}{\partial t} + \frac{\partial (\overline{u_j u_i} + \overline{u_i u_j})}{\partial x_i} = -\frac{1}{\rho} \left(\frac{\partial \overline{p}}{\partial x_i} + \frac{\partial \overline{\tau_{ij}}}{\partial x_j} \right)$$
(2.10)

$$\frac{\partial \overline{u_i}}{\partial x_i} = 0 \tag{2.11}$$

which is the correlation between the fluctuating velocity components and is known as the Reynolds stress term. The existence of the Reynolds stress means that there is no longer a closed set of equations, that is to say, they contain more variables than there are equations. Closure requires use of some approximations, which usually take the form of prescribing the Reynolds stress tensor and turbulent scalar fluxes in terms of the mean quantities.

2.2 Fluid dynamic models

The turbulent flow is highly unsteady and irregular, being necessary high computational resources for representing all the turbulence from the smallest scale corresponding to the dissipative motions to the largest dimension responsible for the majority of the moment transport in high speed. The turbulence models were developed as alternatives to describing the turbulence based on simplified assumptions, being these models summarized as follow:

2.2.1 Direct Numerical Simulation

In the Direct Numerical Simulation (DNS) the Navier-Stokes equations are solved without averaging or approximation to describe the turbulence, only using numerical discretization, thus the errors can be estimated and controlled. This means that the governing equations are solved directly. In the DNS simulation, all of the motions in the flow are solved capturing small scale turbulence. With the use of this method the computed flow field obtained is equivalent to a single realization of a flow or can be considered as short-duration laboratory experiment Tremblay (2001).

In order to capture all the flow motion, in the DSN method the domain in which the computation is performed must be at least as large as the physical domain to be considered or the largest turbulent eddy. Blazek (2001) found that the number of nodes in 3D is proportional to $\mathbf{Re}^{9/4}$ and the time step is related with the mesh size, making the cost of the required computer time proportional to \mathbf{Re}^{3} , Blazek agreed with Frolich et al (1998) who studied the required resolution for DNS. Thus, it is still not practical to solve accurately the non-linear nature and three dimensional characteristics of turbulence at high speed flow using DNS, since the number of grid points to be used is limited by the computational resources.

The DNS it possible to use only for flows at relatively low Reynolds numbers and in geometrically simple domains, being not practical in engineering problems.

2.2.2 Large Eddy Simulation

The Large Eddy Simulation (LES) is just an approach and is classified as a space filtering method in CFD. The LES directly computes large-scale turbulent structures which are responsible for the transfer of energy and momentum in a flow, modelling the smaller scale of dissipative and isotropic structures.

For distinguishing between large and small scales, a filter function is used in LES. A filter function dictates which eddies are large by introducing a length scale, usually denoted as Δ in LES, in the characteristic filter width of the simulation. All eddies larger than Δ are resolved directly, while those smaller are approximated (Feguizer, 2002).

The simulations by LES method are three dimensional, time dependent and expensive, but cheaper than DNS at the same flow. LES is the preferred method for flows in which the Reynolds number is high or the geometry is complex to allow application of DNS.

The main problem of LES is the simulation of the near wall region. Close to the walls the size of the turbulent structures becomes very small. In order to have a well resolved LES it is necessary to have a very fine mesh in the near wall region in order to be able to capture those structures. This makes that in practice the near wall grid size should be almost as small as in DNS. This requirement precludes the use of fully LES for industrial applications at the present time. Spalart (2000) estimates that until later than 2050 the computer's power will not be enough to apply fully LES and DNS techniques to hydrodynamics industrial applications. A common solution is to solve the RANS equations near the wall and the LES equations far away from the wall. This approach is called Detached Eddy Simulation (DES), and it is presently more commonly used than LES.

2.2.3 Detached Eddy Simulation

Detached Eddy Simulation (DES) is an hybrid model that combines the RANS model near to the wall and the LES model in the wake region of a flow, being unsteady and irregular motion of flow is usually found. Basically, the DES model uses a turbulent length scale, L_t to select the approach to use during a simulation. The DES starts with the RANSE model at the inlet boundary, the formulation is the same as the standard two equations model without consideration of the length scale used in the dissipation rate computation, being replaced by local grid spacing. If the turbulent length scale is greater than the grid spacing, which is common in regions with large eddies and chaotic flow nature, the LES is activated in the DES formulation (Liaw, 2005).

2.2.4 RANS turbulence models

The most commonly used approach in engineering practice nowadays is to solve the Reynolds Averaged Navier-Stokes equations coupled with a turbulence model. Here, all the turbulent structures occurring in the flow are modelled. The RANS turbulence models are usually derived for fully turbulent flow and their constants are obtained from experimental data on boundary layers or other simple shear flows. While the turbulence models used in LES are only responsible for modelling the sub mesh scales, in RANS they are responsible for modelling the whole turbulence spectrum.

Several kinds of RANS turbulence models exist. The most used in practise are the linear eddy viscosity models, in which the Boussinesq assumption is used to compute the Reynolds stresses from the mean velocity gradients via a linear relation. There are also non-linear eddy viscosity models, in which the Reynolds stresses and the mean velocity gradients are linked by a non-linear relation. None of the eddy viscosity models can be considered clearly superior to the other ones. The most popular RANS model is the $k - \varepsilon$ model of Jones and Launder (1975) (with all its low-Reynolds versions), which was proposed in the early seventies, and it is still widely used in all fluid dynamics areas.

Other popular models are the *Shear Stress Transport* (SST) model, the $k - \omega$ and the *Spalart-Allmaras* models (Spalart and Allamaras, 1994), which are mainly used in the ship hydrodynamics area. New versions of the models are still appearing, and much work is still being done in order to improve the existing models introducing correction terms which account for specific flow conditions (near wall terms, curvature and rotation corrections, anisotropy effects). The fact of the original $k - \varepsilon$ model being one of the most commonly used two-equation models shows that there has not yet appeared any clearly superior model. A possible reason for the similar results given by the different models under some flow conditions was pointed out by Hunt (1990), who considers that the influence of the turbulence model may be smeared in regions where the time scale of the mean flow distortions is smaller than the characteristic turbulent time scale.

In the *Reynolds Stress Turbulence Models* (RSTM), instead of using the Boussinesq assumption, a transport equation is solved for each Reynolds stress. The fact of solving one

equation for each Reynolds stress allows accounting for curvature effects and anisotropy. In the *Algebraic Stress Models* (ASM) the Reynolds stresses are approximated with non-linear algebraic expressions. The ASM can be thought of either as a simplification of the RSTM or as an extension of Boussinesq eddy viscosity models. Nevertheless, the fact that the equations for the Reynolds stresses still contain modelled terms, and the higher complexity of RSTM compared with eddy viscosity models, make the latter be more commonly used in engineering practise.

2.2.5 Unsteady RANS turbulence model

The purpose of unsteady RANS (URANS) is to simulate the largest eddies present in the flow and their non-linear interaction. Therefore, URANS solutions are unsteady in time even with steady boundary conditions. Durbin (1995) found that the Reynolds stresses created behind a bluff body by time averaging of URANS are larger than those given by the turbulence model, removing in such a way much responsibility from the model.

In principle, URANS is an intermediate approach between steady RANS and LES. The main difference between URANS and LES is that in LES the eddy viscosity of the sub mesh model depends explicitly on the grid size, while URANS is mesh independent by definition. Nonetheless, there are many facts about URANS simulations that are still not clear, which makes LES/DES a more common approach at the present time.

A 3D-URANS computation is able to produce 3D solutions over 2D geometries, like LES, but they appear to be much more dependent on the span wise size of the domain, which is chosen arbitrarily in the computations (Spalart, 2000). In addition, the accuracy of the results depends on the kind of flow, and the solutions have been found to be quite sensitive to the turbulence model (Travin et al, 2004). Although URANS solutions should be mesh independent by definition, there are some recently formulations (Menter et al, 2003) which reduce the value of the eddy viscosity in some regions of the flow in order to be able to resolve smaller turbulent structures, obtaining in such a way an LES-like behaviour. These facts show that there is not yet a complete understanding of the results given by URANS simulations (Travein et al, 2004).

2.3 Linear eddy viscosity turbulence models

The Boussinesq assumption is the base of all the eddy viscosity models. It relates the Reynolds stresses with the mean velocity gradients via the eddy viscosity as:

$$\overline{u'_{i}u'_{j}} = -2v_{t}\left(\overline{s_{ij}} - \frac{1}{3}\overline{s_{kk}}\delta_{ij}\right) + \frac{2}{3}k\delta_{ij}$$
(2.12)

where u'_i is the fluctuating velocity, v_i is the eddy viscosity, $\overline{s_{ij}}$ is the mean strain-rate tensor, and k is the turbulent kinematic energy defined as $k = \frac{\overline{u'_k u'_k}}{2}$. The evaluation of the eddy viscosity, which is assumed to be isotropic, is left to the turbulence model. For a long time simple turbulence models based on algebraic formulations have been used due to their simplicity and robustness. More sophisticated models exist, which solve one or more transport equations for different turbulent quantities, as the turbulent kinetic energy or the dissipation rate.

2.3.1 Standard Spalart Allmaras model

The *Standard Spalart Allmaras* model is one-equation turbulence model available to calculate the turbulence in unsteady state flow model (Houzeaux, 2002). This model involves an eddy-viscosity variable \tilde{v} , related to the eddy-viscosity v_t by:

$$v_t = f_{vl} \tilde{v} \tag{2.13}$$

where:

$$f_{v1} = \frac{x^3}{x^3 + c_{v1}^{3}}; \ x = \frac{\tilde{v}}{v_t}$$
(2.14)

The transport equation for \tilde{v} is:

$$\frac{\partial \tilde{v}}{\partial t} + U_i \cdot \nabla \tilde{v} = C_{b1} \tilde{S} \tilde{v} + \frac{1}{\sigma} \left[\frac{\partial}{\partial x_j} \left((v_t + \tilde{v}) \frac{\partial \tilde{v}}{\partial x_j} \right) \right] + \frac{C_{b2}}{\sigma} \left(\frac{\partial \tilde{v}}{\partial x_j} \right)^2 - C_{\omega 1} f_{\omega 1} \frac{\tilde{v}^2}{d^2}$$
(2.15)

where d is the short distance to the wall and k is the Von-Karman constant. The constants of the model are given in the set of constant below, where the function f_{ω} is given by:

$$f_{\omega} = g \left[\frac{1 + C_{\omega 3}^{6}}{g^{6} + C_{\omega 3}^{6}} \right]^{\frac{1}{6}}$$
(2.16)

with:

$$g = r + C_{\omega} \left(r^6 - r \right); \ r = \frac{\tilde{v}}{\tilde{S}k^2 d^2}$$
(2.17)

g can take relatively high values, so it is preferable to compute f_{ω} as:

$$f_{\omega} = \left[\frac{1 + C_{\omega3}^{6}}{1 + C_{\omega3}^{6}}\right]^{\frac{1}{6}}$$
(2.18)

the production term, the first term of the right-hand side of equation(2.15) involve the quantity \tilde{S} , which is a function of the magnitude of vorticity S and is given by:

$$\tilde{S} = S + \frac{\tilde{v}}{k^2 d^2} f_{v_2}; \ f_{v_2} = 1 - \frac{x}{1 + x f_{v_1}}, \tag{2.19}$$

where:

$$S = \sqrt{2\Omega(u_{ij})\Omega(u_{ij})}; \ \Omega(u_{ij}) = \frac{1}{2} \left(\frac{\partial u_i}{x_j} - \frac{\partial u_j}{\partial u_i} \right)$$
(2.20)

and finally the constant are:

$$C_{\omega 1} = 0.1335; C_{b2} = 0.622; \sigma = 0.667; C_{\nu 1} = 7.1; k = 0.41;$$
$$C_{\omega 1} = \frac{C_{b1}}{k^2} + \frac{1 + C_{b2}}{\sigma}; C_{\omega 2} = 0.3; C_{\omega 3} = 2.0$$
(2.21)

2.3.2 *Two-equation turbulence model* ($k - \varepsilon$ and $k - \omega$)

The two-equation models, which are less demanding in computer resources than the other turbulence models, are an alternative to solve the RANS equations. These models involve the effect of the transport of turbulence quantities by considering the transfer energy in the flow. The calculation of an empirical length scale from a second transport equation is also involved in the calculation of these turbulence methods. In Computational Fluid Dynamics, $k - \varepsilon$ and $k - \omega$ turbulence model are the most widely used models in this category.

The standard $k - \varepsilon$ turbulence model solves the flow based on the rate of production and dissipation of turbulent flows which are in near-balance in the transfer energy. The dissipation rate, ε of the energy is written as:

$$\varepsilon = \frac{k^{3/2}}{L} \tag{2.22}$$

where k is the kinetic energy of the flow and L is the length scale involved. The $k - \varepsilon$ is related to the turbulent viscosity, v_t based on the Prandtl mixing length model:

$$v_t = \rho C_\mu \frac{k^2}{\varepsilon} \tag{2.23}$$

where C_{μ} is an empirical constant and ρ is the density the flow. Applying this constant to the equations governing fluid flow, the *k* equation of the standard $k - \varepsilon$ model is written as:

$$U_{i} \frac{\partial (\rho U_{j} k)}{\partial x_{j}} = \frac{\partial}{\partial x_{j}} \left(\frac{v_{t}}{\sigma_{\varepsilon}} \frac{\partial k}{\partial x_{j}} \right) + v_{t} \frac{\partial U_{i}}{\partial x_{j}} \left(\frac{\partial U_{i}}{\partial x_{j}} + \frac{\partial U_{j}}{\partial x_{j}} \right) \frac{\partial U_{i}}{\partial x_{j}} - \rho \varepsilon$$
(2.24)

and the ε equation:

$$U_{i}\frac{\partial(\rho U_{j}\varepsilon)}{\partial x_{j}} = \frac{\partial}{\partial x_{j}}\left(\frac{v_{t}}{\sigma_{\varepsilon}}\frac{\partial\varepsilon}{\partial x_{j}}\right) + C_{\varepsilon^{1}}\frac{\varepsilon}{k}v_{t}\frac{\partial U_{i}}{\partial x_{j}}\left(\frac{\partial U_{i}}{\partial x_{j}} + \frac{\partial U_{j}}{\partial x_{j}}\right) - \rho C_{\varepsilon^{2}}\frac{\varepsilon^{2}}{k}$$
(2.25)

and based on extensive examination of a wide range of turbulent flows, the constant parameters used in the equations take the following values:

$$C_{\mu} = 0.09; \ C_{\varepsilon 1} = 1.44; \ C_{\varepsilon 2} = 1.92; \ \sigma_k = 1.0; \ \sigma_{\varepsilon} = 1.3$$
 (2.26)

The Shear Stress Transport (SST) $k - \omega$ model was developed as an alternative to cover the deficiencies of the standard $k - \varepsilon$ model at the walls. The SST $k - \omega$ is similar in structure to the $k - \varepsilon$ model, but the variable ε is replaced by the dissipation rate per unit kinetic energy ω . The k equations in the SST model are written as:

$$U_{i}\frac{\partial(\rho U_{j}k)}{\partial x_{j}} = \frac{\partial}{\partial x_{j}}\left(\frac{v_{t}}{\sigma_{\varepsilon}}\frac{\partial k}{\partial x_{j}}\right) + v_{t}\frac{\partial U_{i}}{\partial x_{j}}\left(\frac{\partial U_{i}}{\partial x_{j}} + \frac{\partial U_{j}}{\partial x_{j}}\right)\frac{\partial U_{i}}{\partial x_{j}} - \rho k\omega$$
(2.27)

and the ω equation:

$$U_{i}\frac{\partial(\rho U_{j}\varepsilon)}{\partial x_{j}} = \frac{\partial}{\partial x_{j}}\left(\frac{v_{t}}{\sigma_{\varepsilon}}\frac{\partial\omega}{\partial x_{j}}\right) + \alpha\frac{\omega}{k}v_{t}\frac{\partial U_{i}}{\partial x_{j}}\left(\frac{\partial U_{i}}{\partial x_{j}} + \frac{\partial U_{j}}{\partial x_{j}}\right) - \beta\rho\omega^{2}$$
(2.28)

where:

$$v_t = \rho \frac{k}{\omega} \tag{2.29}$$

Although the two equations models $(k - \varepsilon \text{ and } k - \omega)$ provide a good compromise between complexity and accuracy among RANS models, the applications are restricted to a steady state flow. Thus, solution is sought to achieve both computational efficiency and capability of predicting the irregular nature of flow such as vortex shedding.

2.4 Boundary conditions

For the equations described in section 2.1.1, it is necessary to indicate a series of Boundary Conditions (BC), for defining the physical problem of the fluid flow. The BC in the present work are described below and are applicable to study the interaction phenomenon.

The physical BC deal with the wall of the domain, indicating the type of fluid, which can be a viscous fluid or assumed as an inviscid fluid, using the No-Slip or slip condition respectively (Anderson, 1995).

The No-Slip BC on a surface assumes zero relative velocity between the surface and the fluid flow immediately at the surface. If the surface is stationary, with the fluid moving and passing through, the condition is writing as follow:

$$u_{iwall} = 0 \tag{2.30}$$

where u_{iwall} is the tangential velocity at the surfaces.

The Slip BC for ideal flow considers that there is no friction interacting with the surface boundaries. Hence, the normal velocity of the wall is set to zero and the tangential velocity is a finite non-zero value. If n is a unit normal vector at the point on the surface, the wall boundary condition is given as:

$$\vec{u}_i \cdot n = 0 \tag{2.31}$$

and the term of stress tensor is:

$$\tau_{ii} = 0 \tag{2.32}$$

The governing equation requires that the boundary flows be expressed in terms of known quantities and interior values. It is necessary to indicate the inlet and outlet condition of the flow and the calculation can be simplified with a symmetry condition is possible (Ferziger, 2002).

In the inlet BC, the magnitude of the velocity can be specified in the normal direction to the boundary. The direction constraint requires that the flow direction is parallel to the boundary surface normal, which is calculated at the surface on the inlet boundary. The boundary velocity components are specified, with a non-zero resultant into the domain in Cartesian way and can be written as:

$$u_{inlet} = u_{ispc} \tag{2.33}$$

where u_{iesp} is the specified velocity

In the outlet BC, in general relative static pressure is specified over the outlet boundary by:

$$p_{stacoutlet} = p_{spec} \tag{2.34}$$

being in general the p_{spec} equal to the hydrostatics pressure.

The symmetry plane boundary condition imposes constraints which mirror of the flow on either side. For example, the normal velocity component at the symmetry plane boundary is set to zero:

$$\vec{u}_i \cdot n = 0 \tag{2.35}$$

and the scalar variable gradients normal to the boundary are also set to zero:

$$\frac{\partial \phi}{\partial n} = 0$$
 (2.36)

where ϕ is a general scalar variable.

2.5 Computational Fluid Dynamics solver

When the fluid flow problems are numerically solved, the surfaces, BC and spaces around the boundaries of the computational domain are discretized to be used in Computational Fluid Dynamics (CFD) code. Typical CFD software contains three main modules: the pre-processor, the solver and the post-processor.

Pre-processor: In the pre-processor are included all variables that define the problem setup. In the region of fluid to be analysed it must be defined the properties of the fluid acting on the domain, including external constraints or boundary conditions, such as pressure and velocity to implement realistic situations.

Solver: Here the solution to a CFD problem is computed. The governing equations are solved iteratively to compute the flow parameters of the fluid as the time lapses. Convergence is important to produce an accurate solution of the partial differential equations.

Post-processor: This module is used to process and visualize the results obtained from the solver.

To obtain an approximate solution of the governing equations, a discretized method is used, reviewed in the next section, which approximates the original differential equations with a system of algebraic equations, solved by the CFD software. Discretization in space and time must be defined. The accuracy of numerical solutions depends on the quality of the discretization used.

Spatial discretization divides the computational domain into small sub-domains where the mesh is generated. The fluid flow is described mathematically by specifying its velocity at all points in space and time. Meshes in CFD comprise nodes at which flow parameters are resolved. The three main types of meshes commonly used in computational modelling are: structured (Hexahedral mesh), Unstructured (Tetrahedral mesh), and multi-block structured mesh. The former is used in general for simple geometry such as square or rectangular section shape. The unstructured mesh is used to complex computational boundaries or geometries, thus being discretized, even though require a higher computational cost. The later multi-block mesh is more complicated than the previous one. However combines the advantage of both, structured and unstructured, obtaining a better approach of the space (Liaw, 2005).

The discretization of the time is represented by discrete time steps in the continuous flow. In time-dependent formulations (unsteady state), there is an additional time variable t in the governing equations compared to the steady state analysis. This leads to a system of partial differential equations in time, which comprise unknown terms in the governing equation, at a given time as a function of the variables of the previous time step. Thus, unsteady simulation normally requires longer computational time compared to a steady case due to the smaller time step used to solve de equations.

Either explicit or implicit method can be used for unsteady time-dependent calculation. In an explicit calculation, a forward difference in time is taken when calculating the following time t^{n+1} by using the previous time step value (*n* denotes state at time *t* and n+1 at time $t + \Delta t$) (Stoesser, 2001). The explicit method is straight forward, but each time step has to be kept minimum to maintain computation stability and convergence. On the other hand, the implicit method computes values of the time step t^{n+1} at the same time level in a simulation at different nodes based on a backward difference method. This results in a larger

system of linear equations where unknown values at time step t^{n+1} have to be solved simultaneously.

The principal advantage of implicit schemes compared to explicit ones is that significantly larger time steps can be used, whilst maintaining the stability of the time integration process (Blazek, 2001). A smaller time step Δt in an explicit method implies longer computational running time but it is more accurate.

One important aspect to consider in the calculation is the residual of the solutions. The equations describing fluid flow are solved iteratively obtaining residual values give the accuracy of the result. In engineering application, a residual value is usually situated between four to six orders of magnitude of the actual values (Stangroom, 2001) to achieve convergence of the solution to an acceptable level.

2.6 Numerical methods

In CFD, equation discretization is usually performed by using the following three methods independently: Finite Difference (FD) method, the Finite Element (FE) method and the Finite Volume (FV) method.

2.6.1 Finite Difference method

The Finite Difference (FD) method is the oldest one. It was first developed by Euler in 1768, and is used to obtain numerical solutions to differential equations with simple geometries that cannot be solved analytically. In the method, at each node point of the mesh the Reynolds-averaged Navier–Stokes (RANS) equation is approximated by replacing the partial derivatives by finite difference in terms of the nodal values of the functions (The Taylor series expansions or polynomial fitting is used to obtain the first and second derivative of the variable, see Stroud 1996). The result is one algebraic equation per mesh node, where the variable value at that and a certain number of neighbour nodes appear as unknowns (Feguizer, 2002).

The FD method is the simplest method to apply, but requires a high degree of regularity of the mesh. The method can be used for any mesh type, but in general is used for a structured mesh, because of its simplicity and effectiveness. The node points should form an array in three dimensions, allowing the finite difference approximations to be formed from local and, easily addressed locations.

2.6.2 Finite Element method

The Finite Element (FE) method was developed initially as a procedure for constructing matrix solutions for stress and displacement calculations in structural analysis and later has been adapted in CFD solver. Here the domain is divided into a set of discrete volumes or finite elements that are generally unstructured; in 2D are used triangles or quadrilaterals, whereas in 3D are most often used tetrahedral or hexahedral mesh.

The FE method uses simple piecewise polynomial functions on local elements to describe the variations of the unknown flow variables. When these approximate functions are substituted into the governing equation, the concept of residual values is introduced to measure the errors. These residuals values are minimized by multiplying by a set of weighting functions and then integrating. This results in a set of algebraic equations for the unknown terms of the approximating functions of each node and hence the flow solution can be found (Feguizer, 2002).

The methods are not used extensively in CFD, although there are a number of commercial and research based codes available. For certain classes of flow FE methods bring a high degree of formalized accuracy to the numerical modelling process. However, it has generally been found that FE methods require greater computational resources effort than equivalent Finite Volume methods.

2.6.3 Finite Volume method

The Finite Volume (FV) method is similar to the FE method described above. The FV uses the integral form of the conservation equations as its starting point and the solution domain is subdivided into a finite number of contiguous control volumes. The governing equations of fluid flow are integrated and solved iteratively based on the conservation laws on each control volume. The discretization process results in a set of algebraic equations that resolve the variables at a specified finite number of points within the control volumes using an integration method. Through the integration on the control volumes, the flow around the domain can be fully modelled. The FV method can be used both for the structured and unstructured meshes. Since this method involves direct integration, it is more efficient and easier to program in terms of CFD codes. Hence, FV method has been more recently used in CFD applications than the FE or FD method.

The conservation equations described in section 2.1 can be discretized using an element-based finite volume method. The mesh may consist of tetrahedral, prismatic,

pyramid, and hexahedral elements. Integration point quantities, such as pressure and velocity gradients, are obtained from node values using finite element shape functions, with the exception of advected variables, which are obtained using an upwind-based discretization described by Pertila and Trif (2005).

Hence, the discretization of the conservation Equations (2.1) and (2.2) at each control volume can consider which is fully conservative and implicit. The conservation equations are integrated over each control volume, and the volume integrals are converted to surface integrals using Gauss' divergence theorem (see Altintas, 1990). The discrete representation of Equation (2.1) evaluated with the volume fraction is:

$$\sum_{ip} \left(\rho_q u_i A_{i,ip} \right)_{ip} \left(\alpha_{q,ip} \right)^{n+1} = 0$$
(2.37)

where, $A_{i,ip}$ is the area vector of a sub face corresponding to an integration point, the superscripts n+1 and n mean that the quantity is evaluated at the new and old time step, respectively, in a time interval, Δt .

The advection scheme used to evaluate $\alpha_{q,ip}$ in terms of contiguous vertex values must give solutions which are both bounded and accurate. These expressions are written:

$$\alpha_{q,ip} = \alpha_{q,ip} + \beta \frac{\partial \alpha_q}{x_j} \cdot \vec{R}$$
(2.38)

where $\alpha_{q,ip}$ is the upwind vertex value and \vec{R} is the vector from the upwind vertex to the integration point. A bounded high-resolution scheme can be obtained by making β as close to one as possible, but reducing it where necessary to prevent overshoots and undershoots. The calculation procedure is similar to that described by Barth and Jespersen (1989).

This high-resolution scheme described by Pertila and Trif (2005), gives good advection accuracy when modelling most flows. However, for free surface applications it is still overly diffusive when applied to $\alpha_{q,ip}$ in Equation (2.37). In this situation a compressive scheme is introduced by allowing $\beta > 1$, but boundedness is still maintained by reducing it as much as necessary to prevent overshoots and undershoot. The bounding algorithm is the same as that for the high-resolution scheme described above. The compressiveness of this scheme arises from the fact that it is antidiffusive for $\beta > 1$. It should be that this scheme does not rely on small time steps to obtain its compressiveness, and is therefore equally applicable to steady

state and transient problems. Examples of its compressive characteristics are described by Zwart (2005).

The mass flows must be discretized in a careful manner to avoid pressure–velocity decoupling. This is performed by generalizing the interpolation scheme proposed by Rhie and Chow (1983) for the unstructured mesh-based method used here. Due to that, the equation (2.37) is fully implicit, and therefore it involves the product of implicit variables u_i and α_q at time level n + 1. This implicit product is linearized as follows:

$$\left(u_{i}\alpha_{q}\right)^{n+1} \approx (u_{i})^{n+1}\alpha_{q}^{n} + (u_{i})^{n}\alpha_{q}^{n+1} - (u_{i})^{n}\alpha_{q}^{n}$$
(2.39)

The discrete representation of eq. (2.2) is:

$$\frac{\rho V}{\varDelta t} \left(\left(u_i \right)^{n+1} - \left(u_i \right)^n \right) + \sum_{ip} \left(\rho u_i A_j \right)^{n+1} \left(u_i \right)^{n+1} = -\sum_{ip} p_{ip}^{n+1} A_i + \rho^{n+1} g^i V + \sum_{ip} \left(\left(\tau_{ij} \right)^{n+1} A_j \right)_{ip}$$
(2.40)

where V represent the volume of control.

As with the volume fraction equation, a first-order transient scheme is used as a means of under-relaxing the solution in a physical manner; for time-accurate simulations, a second order scheme is used instead. For the advected velocity, a standard second-order or higher solution scheme is used. Note also that the buoyancy term is fully implicit and, therefore, it introduces coefficients on the volume fraction in the coupled matrix system.

For the volume continuity constraint, the discrete representation of Eq. (2.3) is:

$$\sum_{q=1}^{N} \alpha_q^{n+1} = 1 \tag{2.41}$$

The set of algebraic equations (2.38), (2.39), and (2.41) represent equations for the volume fraction, velocity, and pressure fields, respectively. With two phases, these equations form a 6 x 6-coupled system of equations at each control volume. These equations are solved simultaneously, leading to a fully coupled algorithm. Retaining the variable coupling is a key component of a scalable solution algorithm. "Scalable" means that the solution cost increases linearly with grid size. It must be noted that additional transport equations, such as turbulence, are not implicitly coupled with the mass and momentum system, because they involve weaker inter-equation couplings.

CHAPTER 3 Application of the numerical method

In the previous chapters the hydrodynamic interaction between ships and the Computational Fluid Dynamics (CFD) method were described to define the appropriate conditions and numerical method to apply in the governing equations used in the present work.

Definition of the problem to be analyzed by CFD is presented in this chapter. Different parameters that have influence on the calculation of the interaction forces and moments are specified as well as the characteristics of the ships model establishing the configuration of the mesh and the computational criteria and boundary conditions that are applied in the all CFD simulation software.

3.1 Study of the sensitivity of results to the mesh resolution

Due to computational resource limitations comparative computations were carried out using different mesh size on the ship hull and free surface in order to explore the influence of different mesh size and local refinement on the observed accuracy of the results.

3.1.1 Hull form

The ship model used for this study is a series 60 with block coefficient, C_B of 0.6, which is a single-propeller merchant type ship and is a standard for ship hydrodynamics research, and widely used in ITTC research program. The characteristics of based hull in model scale used for the experimental and computational test are given bellow, and the longitudinal profile of the 3D model is shown in Figure 3.1.

Length between perpendicular	L_{pp}	7.000 (m)
Breadth	В	0.933 (m)
Draft	Т	0.373 (m)
Displacement	∇	1.462 (m ³)
Wetted Surface Area	S	8.349 (m ²)



Figure 3.1 Longitudinal profile of the 3D model.

3.1.2 Experimental set up

To compare and validate the numerical results, use has been made of the experimental results presented by the ITTC Cooperative Experiment on a Series 60 Model, at the Ship Research Institute in the study "Flow Measurement and Resistance Test" (Takeshi et al. 1987).

In the experiments, the resistance test was carried out under free condition. The range of Froude number \mathbf{Fn} was 0.07 to 0.34 and its step is 0.01. The resistance force was measured by a resistance dynamometer of the strain gauge type which has the capacity of 20kg and a tolerance of 0.05% of the full scale. The wave profiles along the hull surface were measured by photographs at values of \mathbf{Fn} of 0.18; 0.22; 0.25; 0.28; 0.30; 0.32 and 0.34. The horizontal and vertical scales were drawn on the model surface for this purpose. The photographs were taken by the three 35mm cameras.

The viscous flow field was measured using 5-hole Pitot tube, which is the NPL type (apex angle is 100deg.) and its diameter is 5mm. The ship model was fixed to the towing carriage in order to assure the accuracy of measuring position. The Froude number was set to 0.18, for calibration. The water conditions in the towing tank are shown in Table 3.1.

Table 3.1 Environment condition measured in the Towing tank.

T°	21.50	°C
ρ	997	Kg/m ³
ν	$0.963 \ 10^{-6}$	m ² /s

The obtained experimental results are shown below together with the results of computations.

3.1.3 Criteria for selecting the mesh

In CFD analyses the flow domain is subdivided into a large number of computational cells. The number of cells in the mesh should be sufficiently large for an adequate representation of the geometry and the phenomena in the flow domain.

A mesh of quality is essential for performing a reliable CFD analysis. Thus, previous analyses of the mesh for large and complex CFD study are necessary. Most of the mesh generators and CFD solvers check the mesh on some parameters, such as the aspect ratio, internal angle, face warpage, right handiness, negative volumes, cracks, and tetrahedral quality (ANSYS, 2004). Some recommendations for grid generation are summarized as follow: a) Avoid high grid stretching ratios: Aspect ratios, should be between 20 to 50 in regions away of the boundary. Aspect ratios may be larger than 50 in unimportant regions. Aspect ratios should not be larger than 20 in the boundary layers. For well resolved boundary layers at high **Re** numbers, the near-wall aspect ratios can be of the order of $10^{5-}10^{6}$

b) Avoid jumps in grid density: Growth factors should be smaller than 1.3.

c) Use a finer and more regular mesh in critical regions, e.g. regions with high gradients of pressure or large changes of the velocity flow.

d) Avoid the arbitrary grid interfaces, meshes finer, or changes in the type element in critical regions. An arbitrary grid interface occurs when there is no one-to-one correspondence between the cell faces on both sides of a common interface, or between adjacent mesh parts.

e) If possible, determine the size of the cells adjacent to wall boundaries in case of turbulence models and before the grid generation.

 f) Numerical diffusion is high when computational cells are not orthogonal to the fluid flow. Thus, they must be avoid

g) The quality of the mesh should be checked by the mesh generator (aspect ratio, internal angle, face warpage, right handiness, negative volumes, cracks, and tetrahedral quality).

h) In general CFD methods allow the application of grid adaptation procedures, where the grid is refined in critical regions (high truncation errors, large solution gradients, et cetera). Thus, the selection of appropriate indicator functions for the adaptation is essential for the success of the simulations. They should be based on the most important flow features to be computed.

i) As a general rule, any important shear layer in the model should be resolved with at least 10 nodes normal to the layer. This is a requirement that requires the use of grids aligned with the shear layers.

3.1.4 Mesh analysis

The geometry of the hull and the volume of control of the grid were obtained in appropriate external software, and the modelled offsets of the transverse section of the model were obtained from the polynomial definition of the cross section for series 60. The modelled surface of the hull was compared with the original one; the difference was only 0.2%, and thus it was considered that the surface was modelled appropriately by the generated mesh.

The volume of control was chosen to be of box shape. The height of the computational domain is $0.25L_{pp}$ and its width is taken to be of $1.5L_{pp}$ due to the symmetry of the problem. The domain inlet boundary is at a distance of $1.5L_{pp}$ ahead of the ship, while the outlet boundary is located at $2.6L_{pp}$ from the ship stern.

The meshes were generated in ICEM CFD using the option low transition to refine the mesh gradually in the zone of interest until a remote zone of the domain. Unstructured tetrahedral grid was chosen in the domain and walls, for free surface and hull surface a prismatic layer mesh (inflation mesh) was applied with an exponential increment between layers. The initial height and the number of layers were determined by the follow parameters. In the free surface total layer height in the interface zone was twice the draft of the hull, and the total thickness around to hull approximately equal to one quarter of draft. The mesh size for the hull and free surface are summarized in Tables 3.2 and 3.3 and the mesh generated are shown in Figs 3.2 and 3.3.

Table 3.2 Different size meshes used in the hull and free surface (F.S.).

Mesh N°	Hull Size	F.S. Size	Domain N° Node
Mesh N°1	$2\% L_{_{pp}}$	$2\% L_{pp}$	425,940
Mesh N°2	$0.5\%L_{_{pp}}$	$2\% L_{pp}$	596,313
Mesh N°3	$0.5\%L_{_{pp}}$	$1\% L_{_{pp}}$	1,874,160
Mesh N°4	$0.25\% L_{_{pp}}$	$1\% L_{pp}$	2,391,549

Table 3.3 Parameters for prism layer mesh applied in the hull and free surface.

Item	N° Layers	Initial height
Hull	20	0.035
F. S.	20	0.001

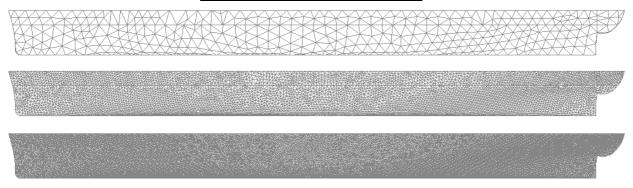


Figure 3.2 Computational mesh on Series 60 surface, at different size mesh.

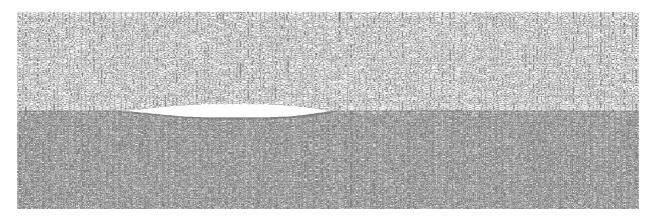


Figure 3.3 Computational grid on water surface around Series 60 ship model, at different size mesh.

The boundary conditions used are summarised as follow: For free surface calculations, the air and water flow around the series 60 ship model using Volume of Fluid (VOF) model available in CFX (ANSYS, 2004). The standard $k - \varepsilon$ and shear stress transport $k - \omega$ turbulent model were employed in the CFD simulations with the standard coefficients. Both turbulence models are widely used in the marine hydrodynamics application and these models have a good performance for high accuracy boundary layer simulations. A velocity inlet boundary condition was used upstream; the flow velocity was considered equal to the velocity experimental. A hydrostatic pressure outlet boundary condition was used downstream; the hydrostatic pressure at the outlet was calculated assuming an undisturbed free surface. Smooth walls with a free-slip condition were assumed for the top, floor and the side wall, only half of the model was considered in the simulations by using a symmetry plane condition at Y = 0. Smooth walls with a non-slip condition (u, v, w = 0) were assumed in the entire hull. The sink and trim of the hull was not taken into account, thus it was considered that the hull was fixed.

The results obtained for different mesh are shown in Figs. 3.4 to 3.7. It was observed that the use of prismatic mesh improves the prediction of the turbulent boundary layer, where the SST model was the best approximation. In higher Froude Number, the predicted resistance is more exact, defining a velocity range where the CFD code gives accurate results without requiring a great computational cost. For the lower Froude Number it is necessary to use a fine mesh capable to consider small differences in the gradient of pressure and free surface deformation. The fine mesh in the interface of the fluid permits a good approximation of the wave pattern by the ship. However the fine mesh in the hull improves the prediction of the resistance better than the fine mesh in the free surface.

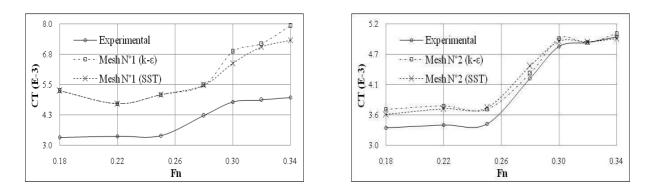


Figure 3.4 Total coefficients at mesh N°1 and N°2 for $k - \varepsilon$ and $k - \omega$ turbulent model, for different **Fn**.

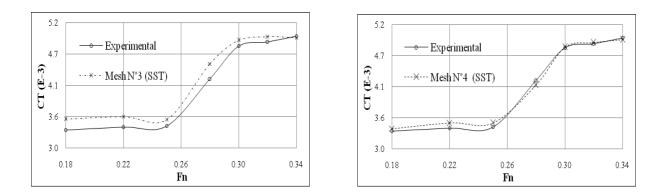


Figure 3.5 Total coefficients at mesh N°3 and N°4 for SST turbulent models for different ${f Fn}$.

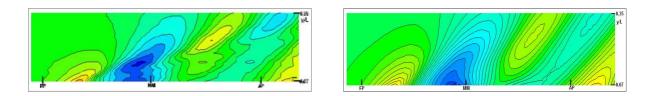


Figure 3.6 Experimental and predicted wave contour for mesh N°2, at Fn = 0.32.

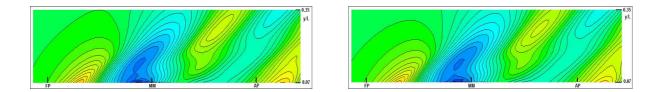


Figure 3.7 Predicted wave contour for mesh N°3 and N°4, at $\mathbf{Fn} = 0.32$.

3.2 Problem Statement: Hydrodynamic interaction between ships

The numerical study was organized to match experimental data on interaction between two ships, a tug and a tanker. Ships are interacting at different positions sailing at the same velocity and in parallel course. Different cases were studied for the ship interaction, the set of velocity are summarized in Table 3.4, where $\mathbf{Fn} = 0.32$ is defined as:

$$\mathbf{Fn} = \frac{U}{\sqrt{gL_{ug}}} \tag{3.1}$$

and **Re** is defined as:

$$\mathbf{Re} = \frac{L_{iug}U}{v}$$
(3.2)

Additional simulations with the tug model sailing freely were performed, being the interaction force obtained as the difference between the values in presence and absence of the tanker model, the interaction forces and moments were obtained by the following expressions:

$$X_i = X_e - X_p \tag{3.3}$$

$$Y_i = Y_e - Y_p \tag{3.4}$$

$$N_i = N_e - N_p \tag{3.5}$$

where $X_{e,p}$, $Y_{e,p}$ and $N_{e,p}$ are the horizontal component of the proper hydrodynamics inertial forces on the tug whit the presence and the absence of the tanker, respectively, and X_i , Y_i and N_i are the pure interaction forces.

Table 3.4 Set of the velocity

U(Kn)	Fn	Re
4	0.121	4.82 E+7
6	0.181	7.22 E+7

During the numerical study varied were the side distance η and the longitudinal offset ξ of the tug with respect to the tanker, and the numerical values for all parameters are given in Tables 3.5, and 3.6. The following non-dimensional parameters are used to represent the results:

$$x' = \frac{2\xi}{L_{\text{tanker}}}$$
(3.6)

$$y' = \frac{2\eta}{B_{\text{tanker}}}$$
(3.7)

For the force coefficients the expressions are:

$$X' = \frac{2Xi}{\rho A U^2} \tag{3.8}$$

$$Y' = \frac{2Yi}{\rho A U^2} \tag{3.9}$$

$$N' = \frac{2Ni}{\rho A L_{tug} U^2} \tag{3.10}$$

where:

$$A = \sqrt[3]{\nabla_{tug} \nabla_{tan \, ker}} \tag{3.11}$$

The studies are performed for two shallow water depths denoted by Depth 1 and Depth 2. The Depth 1 and Depth 2 correspond to the depth-draught ratio, H/T of 1.19 and 1.51, respectively.

Table 3.5 Set of longitudinal distance

Fn	0.121	0.181	
ξ (m)	1.3	58.7	

Table 3.6 Set of lateral distance

Fn	η (m)						
0.121	21.5	21.5 22.1 22.5 23.2 - 31.2 36.2					
0.181	21.5	22.1	-	23.2	26.2	31.2	-

3.3 Ships Model

The vessels used for the simulation were modelled in natural scale (1:1) using Rhinoceros V4.0 (Tutorial Rhinoceros). The main dimensions and shape coefficients are summarized below and the hulls forms are sketched in Fig. 3.8 and 3.9.

		tug	tanker
Length between perpendicular	L_{PP}	26.60 (m)	186.0 (m)
Breadth	В	11.00 (m)	31.6 (m)
Draft	Т	4.60 (m)	10.3 (m)
Displacement	∇	650.0 (m ³)	49197 (m ³)
Block Coefficent	$C_{\scriptscriptstyle B}$	0.43 (-)	0.81 (-)

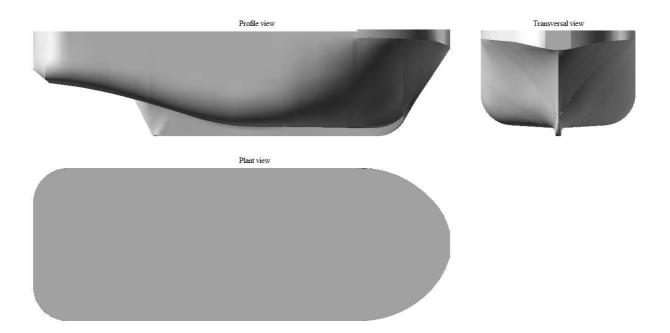
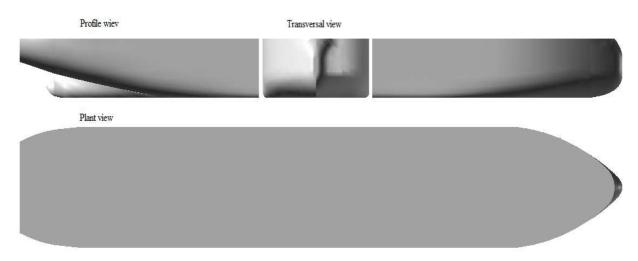
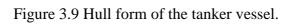


Figure 3.8 Hull form of the tug vessel.





3.4 Simulation Setup

The numerical investigations were carried out by two main CFD models, these models depended on the type of flow (viscous or an inviscid fluid flow), which were performed by two codes. First CFX (ANSYS, 2004) was used, where the simulations were carried out without free surface (Fig. 3.10). After STAR CCM+ (CD-Adapco, 2007) was used taking into a count the free surface deformation (Fig. 3.11).

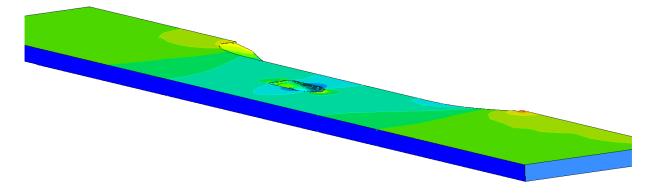


Figure 3.10 Simulation in CFX with waveless.

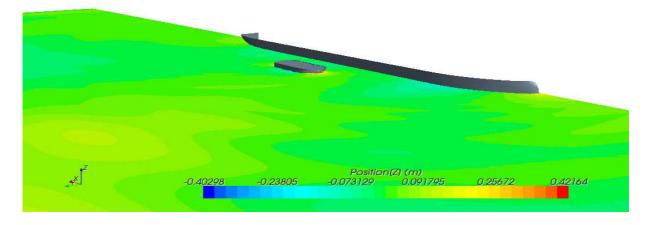


Figure 3.11 Simulation in STAR CCM+ with wave making.

The computations were carried out assuming the flow was unsteady through settling STAR CCM+. To the steady state CFX was chosen for both viscous and inviscid model. For simulating the inviscid Flow in STAR CCM+ was chosen the ideal flow option, while in CFX the viscous of the flow was indicated with the value $v_t = 0$ to simulate the ideal flow.

The *Standard Spalart Allmaras* turbulence model was use in STAR CCM+, while the *Shear Stress Transport* $k - \omega$ model was applied in CFX computation.

The water condition was modelled as fresh water at 21.5°C, ($\rho = 999$ kg/m³, vt= 1.137E-3kg*s/m). The air was assumed compressible (for computational stability reason) and was modelled with a molecular mass of 28.96kg/kmole and a vt = 1.80E-5kg*s/m. The buoyancy forces due to fluid density difference were modelled in the STAR CCM+ simulation.

The total time of the simulation was fixed in STAR CCM+ and equal to 30s using a time step equal to 0.05s with 10 iterations for each step. In CFX the maximum number of iterations was equal to 500 using a time step equal to 0.1 seconds. However, if the CFX convergence criteria described below were reached for all residuals, the simulation was stopped before reaching 500 iterations. For most of simulations, convergence of all residuals forces and monitoring points was achieved in around 300 iterations.

In both programs, the convergence was assessed by plotting the flow parameters against the iteration number these were: residuals for mass, momentum and turbulence (target criteria = 10^{-4}), surge and sway force, and yaw moment.

3.5 Definition of the mesh and Computational Domain

In both simulations, the Volume Control was chosen as a Box Shape, and the dimension was estimated using the proposed by Fonfach and Guedes Soares (2009). The height of the air phase was taken equal to $0.25L_{tanker}$ (the computation with rigid free surface the air phase was not considered), the domain inlet boundary is at a distance of $1.5L_{tanker}$ ahead of the ship, while the outlet boundary is located at $2.6L_{tanker}$ from the ship stern, and its width is taken to be of $1.5L_{tanker}$. Finally the height of the water phase was the water depth defined in section 3.1 for each case.

In the computation with rigid free surface the mesh generator ICEM CFD was used for meshing the computational domain with unstructured tetrahedral grid. Here, only the water phase was meshed. For meshing the volume domain was considered the hulls surface (tug and tanker) with a value $1.0\% L_{tug}$ and $0.5\% L_{tan ker}$, using low transition for refining the mesh gradually in the volume around the ships. For meshing the free surface it was considered like the boundary wall without parameter for fine mesh. For simulating the viscous flow, a prismatic layer mesh was applied in the surface of the tug hull, and thus the height of the layer

was determined based in the previous mesh sensitivity study. The values of the mesh sizes are given in Table 3.7 and the generated meshes are shown in Figs. 3.13 to 3.16.

Part	Tug Hull	Tanker Hull	Other wall	Prims
Falt	(m)	(m)	(m)	layers (m)
CFX Inviscid Model	0.2	1.00	5.00	-
CFX Viscous Model	0.2	1.00	5.00	0.01

Table 3.7 Mesh size for simulation with rigid free surface

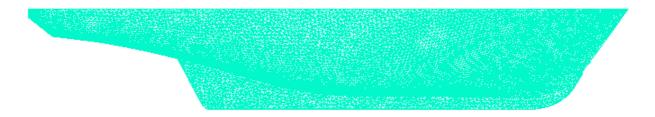


Figure 3.12 Computational mesh on tug hull surface for simulations without wave.

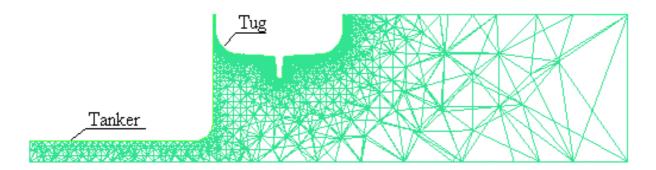


Figure 3.13 Mesh section of the CFX domain for $\mathbf{Fn} = 0.121$ at *x/Ltug* = 0.15.

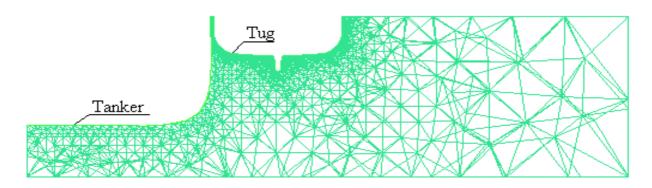


Figure 3.14 Mesh section of the CFX domain for $\mathbf{Fn} = 0.181$ at *x/Ltug* = 0.15.

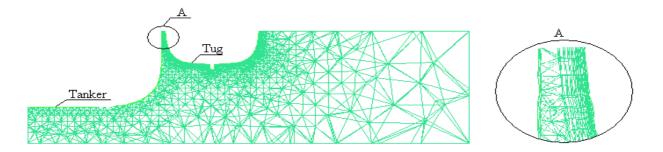


Figure 3.15 Mesh section of the CFX domain and zoom in the prism layer mesh for $\mathbf{Fn} = 0.181$ at x/Ltug = -0.15.

The hexahedral mesh used in the other simulation was generated in STAR CCM+. It should be noticed that in the rigid free surface of the computation for the viscous flow simulation, the prismatic layer mesh was applied around the hull of the tug. The mesh size was determined using the previous mesh sensitivity analysis. The meshes of the hulls and free surface were determined as a percentage of the length of the ship, which gave good result for a large range of Fn. For meshing the surface of the tug hull a mesh size equal to $0.25\% L_{tug}$ was used, whereas for the tanker a mesh size of $1.0\% L_{tan ker}$ was used.

For the interphase of the water and air a block shape was used to refine the free surface zone, and the box height was equal to 1.0m, where z is 0.5m and -0.5m in each fluid, respectively. Three different zones were considered on the free surface for the decrease of the mesh. One of them is the total dimension of the horizontal plane of the volume domain (far away from the both hull) considering a mesh size equal to $0.5\% L_{tan ker}$. Other is half of the width and the total longitudinal dimension of the box (near of the tanker ship) considering a mesh size equal to $0.25\% L_{tan ker}$. Finally using the dimension of the recommendation given by Fonfach and Guedes Soares (2010), taking the length of the tug based for the determined the zone near the tug, and using a mesh size equal to $0.125\% L_{tan ker}$. In Star CCM+ a growing factor value equal to 1 was fixed for a smooth transition between the different cell sizes. The meshes are shown in Figs. 3.17 to 3.20 and the mesh sizes are summarized in Tables 3.8.

Part	Tug Hull (m)	Tanker Hull (m)	F. S. (m)		Other wall (m)	Prism layers (m)	
STAR CCM+ Inviscid Model	0.0725	1	1	0.5	0.25	5	-
STAR CCM+ Viscose Model	0.0725	1	1	0.5	0.25	5	0.01

Table 3.8 Mesh size for simulation with deformable free surface (F.S.)

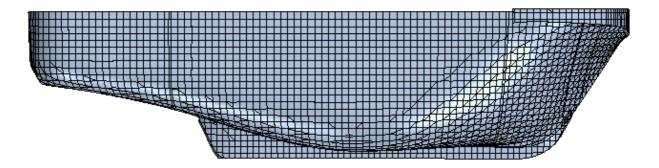


Figure 3.16 Computational mesh on tug hull surface for simulations with wave making

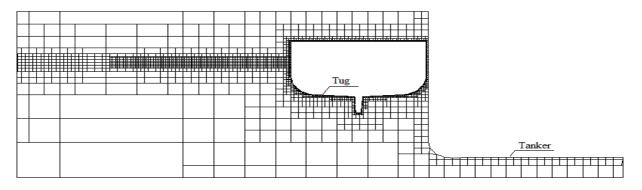


Figure 3.17 Mesh section of the STAR CCM+ domain for $\mathbf{Fn} = 0.121$ at *x/Ltug* = 0.15

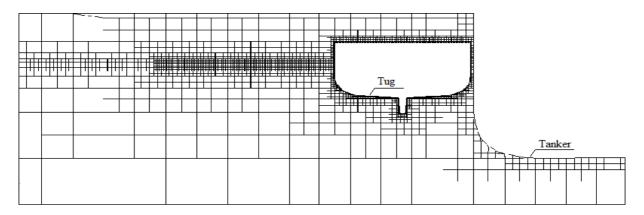


Figure 3.18 Mesh section of the STAR CCM+ domain for $\mathbf{Fn} = 0.181$ at *x/Ltug* = 0.15

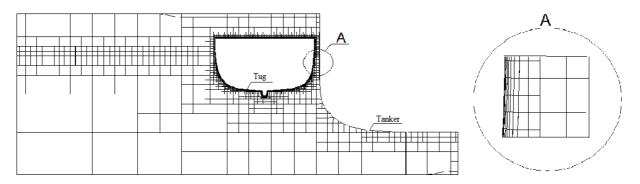


Figure 3.19 Mesh section of the STAR CCM+ domain and zoom in the prism layer mesh for $\mathbf{Fn} = 0.181$ at *x/Ltug* = 0.15.

3.6 Computational Boundary Conditions

The non-zero flow velocity condition was considered for t = 0 (the value of velocity corresponds to each case simulated), with a reference pressure of to 101,325Pa in all simulations. In STAR CCM+ the free surface was considered initially as plane surface, defining a hydrostatic pressure for the water volume. For free surface calculations, the air and water flow around the tug and the tanker models, were simulated using the standard homogenous Volume of Fluid (VOF) model (or free surface model which is available in STAR CCM+ for both inviscid and viscous flow simulation). In the VOF model, a single moment equation is shared by the fluid volume fraction of each of the fluids in each computational cell and is tracked throughout the domain. The surface tension was not applied in the models. The coupled volume fraction was used to improve the convergence. The free surface in CFX was considered as a rigid wall without friction for both ideal and viscous flow simulation

The boundary condition was employed to simulate the experimental condition on the towing tank. An initial velocity boundary condition was used at the beginning of the flow domain in both CFD codes; the flow velocity was considered according to each case studied.

In STAR CCM+ the hydrostatic pressure outlet boundary condition was used at the end of the flow domain; the hydrostatic pressure at the outlet was calculated assuming an undisturbed free surface. While in CFX the outlet hydrostatic pressure was considered p = 0.

Smooth walls with a free-slip condition were assumed for the top, floor and the side wall. Only half of the tank model was considered in the simulations by using a symmetry plane condition at Y = 0, in the other hand the tug was full model. Smooth walls with a non-slip condition (u, v, w = 0) were assumed in both ship hulls in the viscous flow simulation.

CHAPTER 4 Analysis of the numerical results

Results of the numerical analysis carried out using the formulation and techniques described above will be present here.

The description of the interaction coefficients of surge and sway force components and the interaction coefficient of yaw moment acting on the tug are analysed. Comparison between several numerical result and experimental data were also performed.

The summarized the qualitative analysis of the CFD simulation are analysed. This analysis considers the interaction of the wave pattern generated by both ships, the pressure on the tug, and comparison of the velocity around various sections of the tug.

Concerning the tug position with respect to the tanker and the water depth, a distinction is made between:

Case I : Ship to ship navigation at H_1 , **Fn** of 0.121 and x' of +0.014

Case II : Ship to ship navigation at H_2 , **Fn** of 0.181 and x' of +0.61

Additional analyses of the isolated tug were carried out calculating the forces and moments in each case above, to obtain finally the pure interaction forces.

4.1 Isolated tug

The main objective in the calculation of the hydrodynamic coefficient of the isolated tug is to calculate the pure hydrodynamic coefficients for the Cases I and II, considering their respective \mathbf{Fn} .

The hydrodynamic coefficients in Table 4.1 are summarized for $\mathbf{Fn} = 0.121$ and compared with the experimental result. The surge coefficient show good agreement between the computed and the experimental data. Here, the simulations were carried out with deformable free surface to have a better accuracy. A difference of 3.3% for inviscid fluid and 4.5% for the viscous fluid, respect to the experimental was considered. On the other hand, the computations with rigid free surface were less accurate, having a difference of 25% for inviscid fluid and 7% for viscous fluid when compared with the experimental result. In case of the sway coefficient, the computed coefficients were near to zero value, as must be because of the symmetry. Similar analysis for the yaw moment can be adopted.

For $\mathbf{Fn} = 0.181$ the numerical coefficients are summarized in table 4.2. The surge coefficient can be observed to converge. The sway and yaw coefficients are close to zero value were due for the residual value in the computation.

Item		Experimental				
Coefficient	Inviscid Fs	Inviscid Fs Inviscid Sfs Viscous Fs Viscous Sfs				
Χ'	0.0106	0.0087	0.0110	0.0119	0.01095	
Y'	0.0003	0.0007	0.0002	0.0009	0.00363	
N'	0.0001	0.0000	0.0000	0.0000	0.00071	

Table 4.1 Hydrodynamics coefficients for $\mathbf{Fn} = 0.121$

Table 4.2 Hydrodynamics coefficients for $\mathbf{Fn} = 0.181$

Item		Experimental			
Coefficient	Inviscid Fs	Total			
X'	0.0245	0.0256	0.258	0.262	-
Y'	0.0000	0.0002	0.0030	0.0088	-
N'	0.0000	0.0000	0.0013	0.0001	-

The properties of the wave pattern around the hull of the tug and the effects of shallow water on the generated wave at \mathbf{Fn} of 0.121 and 0.181 can be seen from the predicted numerical results in Fig. 4.1. It is observed that the generated wave patterns for both \mathbf{Fn} have similar distribution. A crest at the bow of the tug followed by a valley at the forward side and a pronounced crest of the wave at the mid of the ship, produce a lager valley in the stern side. However, differences are observed between both \mathbf{Fn} . In the first case of \mathbf{Fn} of 0.121, the length of the valley is smooth at the forward side of the tug whereas in the other case of \mathbf{Fn} of 0.181 the valley is pronounced. For \mathbf{Fn} of 0.181 the peak of the crest at the mid-section of the tug is higher at the lower \mathbf{Fn} of 0.181 reaching almost the double high. The valley of the wave at the stern side is deeper and short for higher \mathbf{Fn} .

The deformation of the free surface at the stern of the ship for both \mathbf{Fn} , shows an elevation of the water, being for the first one almost the same at the bow of the tug, whereas the second \mathbf{Fn} the height is higher in stern. This situation is unusual and is due to the shallow water effect. A wake in the flow at the stern of the tug can be noticed due to the robust shape of the hull.

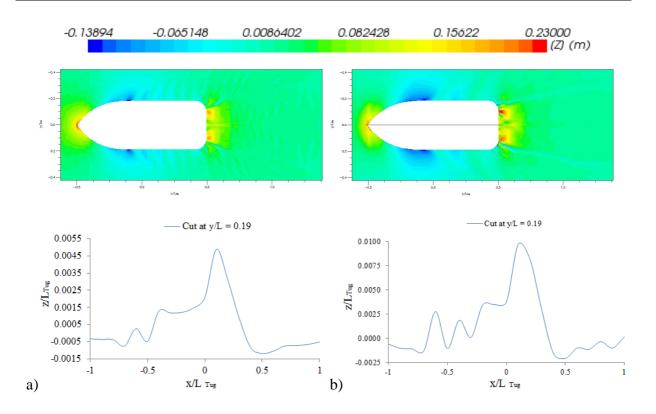


Figure 4.1 Predicted wave patterns by the tug and wave cuts at y/L_{tug} of 0.19 and -0.19: a) Free surface for **Fn** of 0.121; b) Free surface for **Fn** of 0.181.

The distribution of pressure on the Tug hull at both \mathbf{Fn} of 0.121 and 0.181 and calculated by ANSYS CFX and STAR CCM+ are shown in Fig. 4.2 and 4.3. Differences were found in the distribution of the pressure between the model without waves and with wave making (Fig. 4.2). It is noted that the distribution of the pressure on the tug hull is only the hydrodynamic pressure in computation without waves (Fig 4.2a). In computation with wave making the distribution of the pressure on the hull is dominated by the hydrostatic pressure, and the hydrodynamic pressure is smaller (Fig 4.2b).

The same analysis can be adopted for **Fn** of 0.181 (Fig 4.3). The distribution of the pressure on the hull computed without waves are similar for both **Fn**. High pressure is observed at the bow of the tug which decreases between x/L_{tug} between -0.5 to -0.25. Low pressure is produced in the mid of the tug increasing between x/L_{tug} between -0.5 to -0.85, and a high pressure is distributed at the stern of the ship.

The distribution of pressure on the hull computed with deformable free surface for both \mathbf{Fn} shows a similar patter with the maximum high pressure in the keel which decrease going up to the free surface. However, differences are observed, due to the position of the generated wave at the free surface. For \mathbf{Fn} of 0.121 the pressure on the hull is almost constant due to the small deformation of the free surface whereas for the \mathbf{Fn} of 0.181 the high pressure is produced in the bow and is followed by a decreasing when the flow enters in the gap between the ships and is increased again at the mid-section of the tug reaching a constant magnitude close to the stern.

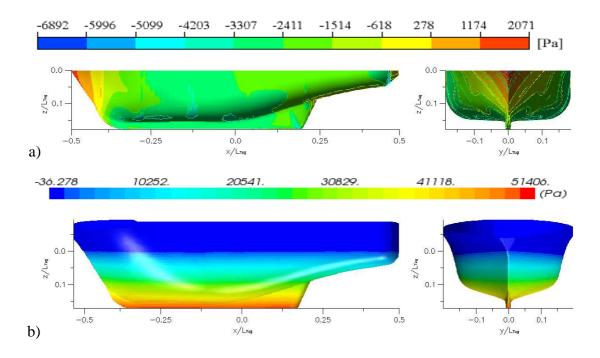


Figure 4.2 Distribution of the pressure on the tug hull surface at **Fn** of 0.121: a) Pressure distribution without wave making; b) Pressure distribution with wave making.

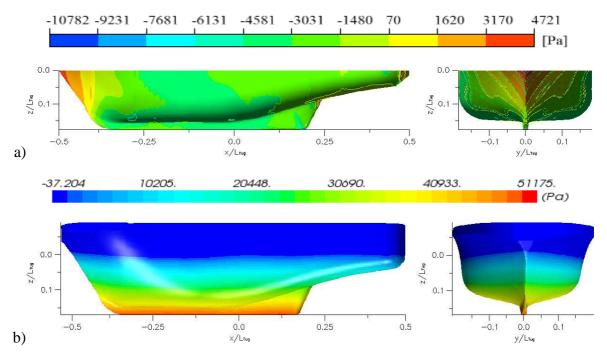


Figure 4.3 Distribution of the pressure on the tug hull surface at **Fn** of 0.181: a) Pressure distribution without wave making; b) Pressure distribution with wave making.

The velocity flow around the tug is plotted in Figs. 4.4 to 4.5. It is observed that both models, with and without wave making show agreement with the predicted velocity for the flow. In Froude numbers of 0.121 and 0.181, it is observed that the velocity decrease around the bow section. However, the velocity increased at the mid-section of the tug reaching the maximum value. The maximum velocity flow was found at the bilges of the tug ship where it was noticed an increment in the velocity. The velocity remains constant at the bilges in the stern zone. The velocity is the same that in the mid ship section of the tug. Below the bottom, the velocity flow decrease to a minimum value (close to zero).

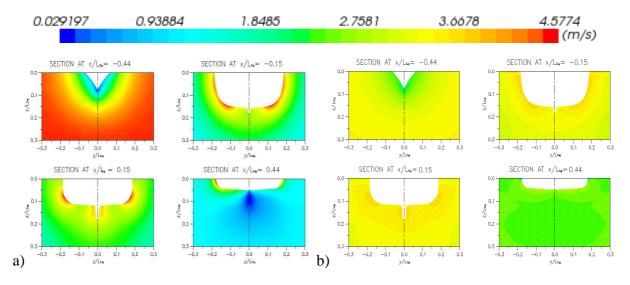


Figure 4.4 Flow velocities distribution around transversal sections of the tug's hull for **Fn** of 0.121: a) Velocity without waves; b) Velocity with wave making.

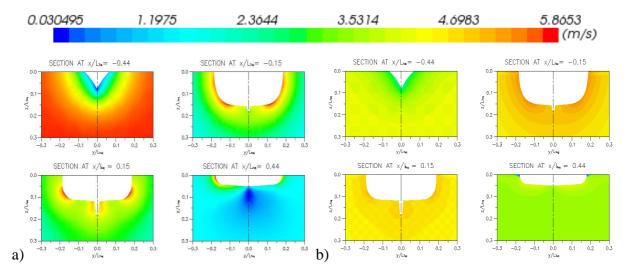


Figure 4.5 Flow velocities distribution around transversal sections of the tug's hull for **Fn** of 0.181: a) Velocity without waves; b) Velocity with wave making.

4.2 Interaction between the ships

4.2.1 Case I (**Fn** of 0.121)

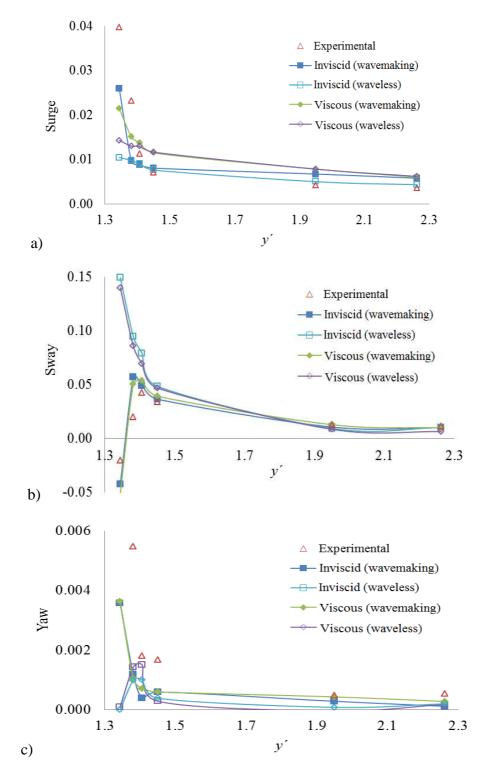
The pure interaction force and moment coefficients with the dimensionless lateral distance between the ships are shown in Fig. 4.6. It is observed that the magnitude of the surge coefficients increases in the negative direction for small gaps. The sway coefficient shows a variation in the direction of the lateral clearance. The tug experiences a repulsion sway force at small gaps between the ships, changing to suction force when the tug vessel increases the lateral distance (between positions y' 1.34 to 1.38). For the remaining the lateral distances, the sway coefficients are in negative direction, being the tug attracted by the tanker vessel. The yaw moment coefficients have important variations, when the ships are widely separated (y' > 1.9) the coefficients are near to zero. On the other hand, for small gaps, the yaw moment coefficients experience large increments.

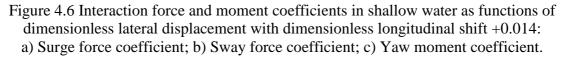
In computations without waves, when the free surface is modelled with a rigid wall all coefficients (without free surface) show good agreement with the experimental results for relative large side clearance. When the tug is very close the tanker this model loses accuracy. This disagreement is critical in the calculation of the sway coefficient at small lateral distance. The yaw moment is smaller than the experimental results. This is noticed clearly at y' = 1.34 where the yaw moment is near to zero.

The simulations carried out with accurate boundary condition on free surface followed the trend of the experimental results. However, even this model presented not very accurate results for the three hydrodynamic coefficients at y' of 1.34 and 1.38. The analysis indicates that for small distances, the accuracy can be improved using an appropriate discretization in the free surface region. It is observed for the larger gaps, both with rigid and deformable free surface are able for predicting the interaction coefficient.

Figs. 4.7 to 4.9 illustrate the computed interaction wave pattern for each lateral distance and along the two planes which is between -0.19 and 0.19m away of the plane of symmetry. In Fig. 4.7a (y' = 1.34) it is possible to observe a pronounced asymmetrical free surface. The waves generated on the interaction side are distributed above the design water line of the tug, while on the free side the waves are produced below the water line. Fig. 4.7b (y' = 1.38) shows the generated wave with the tanker suction over the tug, that means, at the

mid-section of the tug a valley (z/L_{tug}) of -0.018 is produced between the interacting ships while at the free side, the valley is of -0.006.





Figs. 4.8a and 4.8b show the free surfaces for y' 1.41 and 1.46, respectively. It is observed that both lateral distances are close, producing similar wave patterns, as was commented for the lateral gap of 1.38. The interaction between the ships generated a valley at mid-section of the tug in both cases y' 1.41 and 1.46, However, for y' of 1.41 the peak value of the valley are different in both sides of the tug, whereas the y' of 1.46 the wave is almost symmetrical in both sides along of the tug hull.

Here, in these cases of lateral distances the effect of the tanker in the wave pattern produced by the tug can be observed clearly, being the wave patterns generated evidently differently than in the free navigation. In Fig. 4.11 can be seen the free surface for y' of 1.95 (Fig 4.9a) and 2.26 (Fig 4.19b). In these two lateral distances the ships produce symmetric wave patterns because the tug is relatively far away from the tanker, here the valley in the interaction side is less pronounced than in the previous cases, and the amplitude for both y'have the same value in the free side. Considering the cases of lateral distances 1.38 to 2.26, the wave patterns generated by the ships have similar characteristics. It is observed that the wave generated in the bow is slightly higher when the ships interact, and also in the valley generated in the mid-ship section of the tug.

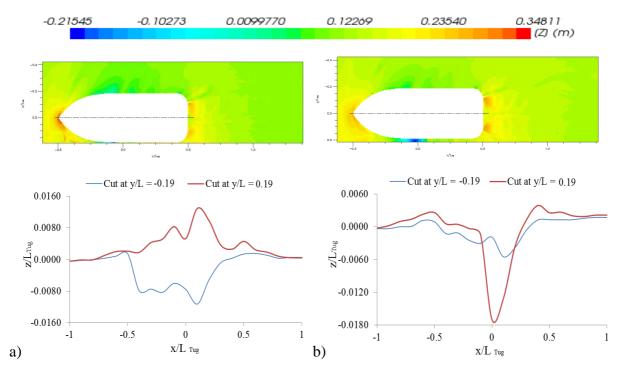


Figure 4.7 Predicted interaction wave patter by the ships for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and wave cut at y/L_{ug} of 0.19 and -0.19: a) Free surface for y' of 1.34; b) Free surface for y' of 1.38.

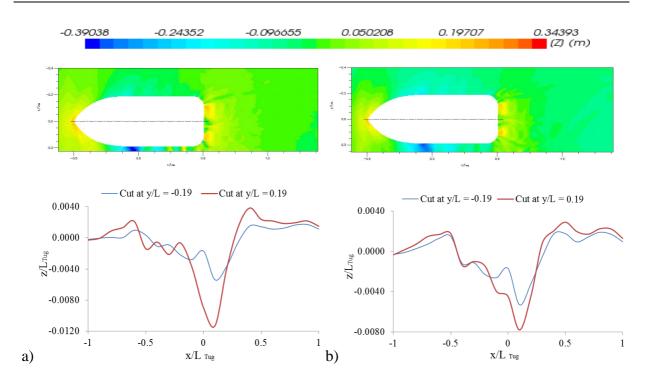


Figure 4.8 Predicted interaction wave patter by the ships for **Fn** of 0.121 with dimensionless longitudinal shift +0.0014 and wave cut at y/L_{mg} of 0.19 and -0.19: a) Free surface for y' of 1.41; b) Free surface for y' of 1.46.

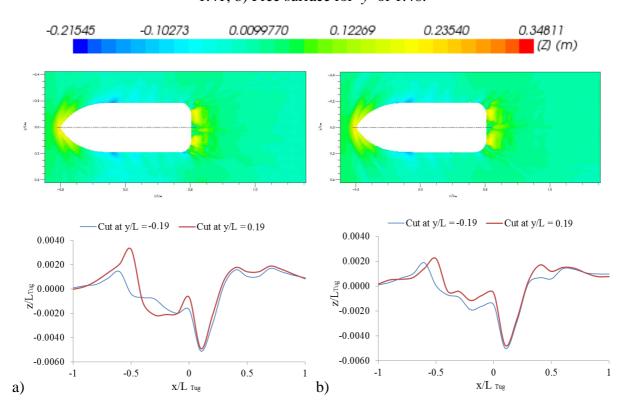


Figure 4.9 Predicted interaction wave patter by the ships for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and wave cut at y/L_{mg} of 0.19 and -0.19: a) Free surface for y' of 1.95; b) Free surface for y' of 2.26.

When y' = 1.34, both solutions (with and without deformable free surface) show differences in the distribution of pressure on the hull, as shown in Fig. 4.10. Fig. 4.10a shows high pressure at the bow that decrease between x/L_{tug} -0.5 to 0.25. A low pressure is observed near to the mid-ship of the tug (between x/L_{tug} -0.25 to 0) producing that the tug ship is suctioned by the tanker ship. In the positions between x/L_{tug} 0.25 to 0.5 the pressure is increased but is smaller than the pressure in the bow. These distributions of the pressure suggest that the position of the sway forces is forward of the mid-ship. In the body view is noted the asymmetric pressure, which is higher at the interaction than at the free side.

The surge force is positioned in the interaction side. It is noticed that the resultant yaw moment on the tug ships is the difference between the moments produced by the two component forces, which is observed in the yaw coefficients (Fig 4.6c). In the Fig. 4.10b (with deformable free surface) is shown the increment of the generated pressure at the bow, compared with the isolated tug. This effect is due to the stationary water in the zone of the curvature of the bow of the tug and at the side of the tanker (between x/L_{tug} -0.5 to 0.25).

This is followed of the small gap between the ships (between x/L_{tug} -0.25 to 0.5) where an increment of the pressure generating a repulsion sway forces can be noted. The small distance between the ships is different of the computation with the rigid free surface. However, this phenomenon is in agreement with the experimental as can be seen in the experimental sway coefficient.

When y' is 1.38 to 2.26 (Figs. 4.10 to 4.15) the distribution of the hull pressure is almost the same in both simulations, the high pressure is observed at the bow of the tug. In the body view is observed that the distribution of the pressure is higher at the interaction than at the free side. The pressure decreased in the aft direction until the mid-section of the ship. The low pressure generated on the tug hull produces a suction sway force. Similar to y' of 1.34, the interaction yaw moment is produced by the difference of the forces due to the sway force, which is forward to the mid ship section whereas the surge force is at the interaction side.

The flow velocities around the transverse sections of the tug hull are shown in Figs. 4.16 to 4.21 where it is observed the influence of the lateral gap along of the tug. When y' of 1.34 (Fig. 4.16), differences between the computations (with and without free deformable free surface) were found at the mid-section going to the stern of the tug. In the calculation with rigid free surface, is observed an increment of the velocity flow in the interaction side. In the

calculation with deformable free surface, the velocity flow decreased flow to zero in the gap between the ships.

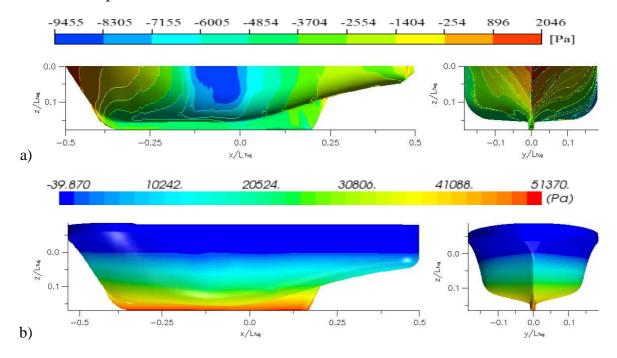


Figure 4.10 Distribution of the pressure on the tug hull surface for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.34: a) Pressure distribution without waves; b) Pressure distribution with wave making.

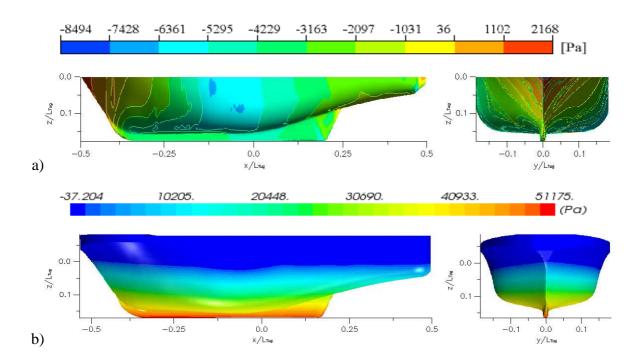


Figure 4.11 Distribution of the pressure on the tug hull surface for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.38: a) Pressure distribution without waves; b) Pressure distribution with wave making.

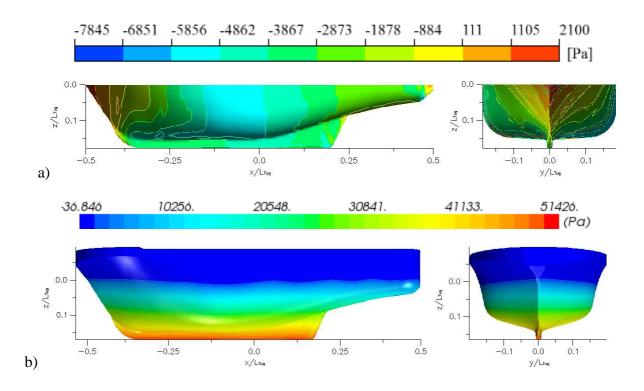


Figure 4.12 Distribution of the pressure on the tug hull surface for **Fn** of 0.121 with dimensionless longitudinal shift +0.0014 and dimensionless lateral shift +1.41: a) Pressure distribution without waves; b) Pressure distribution with wave making.

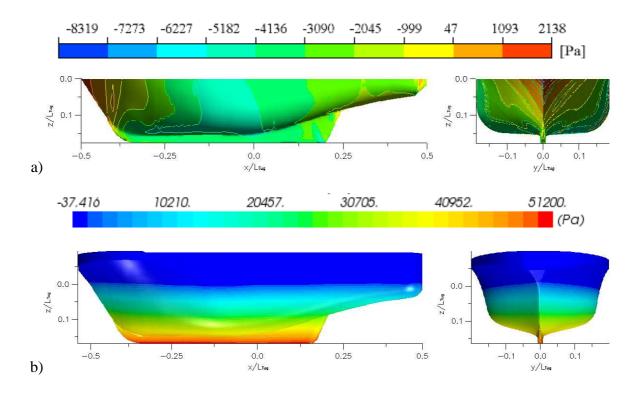


Figure 4.13 Distribution of the pressure on the tug hull surface for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.46: a) Pressure distribution without waves; b) Pressure distribution with wave making.

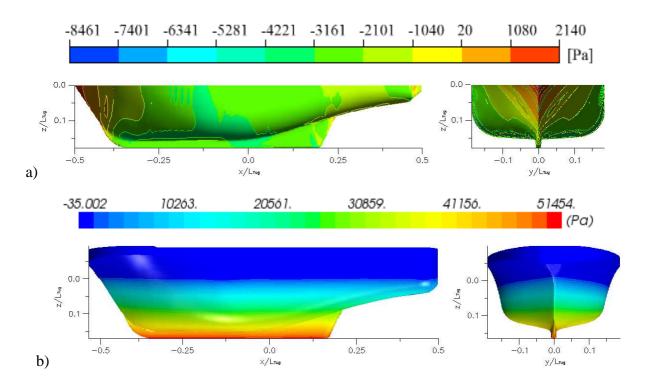


Figure 4.14 Distribution of the pressure on the tug hull surface for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.95: a) Pressure distribution without waves; b) Pressure distribution with wave making.

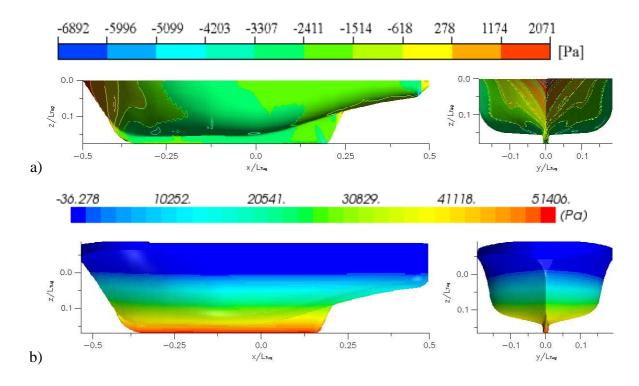


Figure 4.15 Distribution of the pressure on the tug hull surface for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +2.26: a) Pressure distribution without waves; b) Pressure distribution with wave making.

On the stern of the ship, the differences between the models persist. In the simulation without deformable free surface, the velocity decrease in the gap between the ships. However, in the free bilge the velocity increases and is observed a low velocity below the bottom. In the other simulation with deformable free surface, the velocity decrease to zero between the ships. In the rest of the sections the velocity has a constant distribution. In both computations, the velocity of the flow at the keel is asymmetrical, differing with the isolated tug. In the interaction between the ships, the velocity increase more at the side with interaction than in the free side. In the bow of the tug (between x/L_{tug} -0.15 and -0.44) both computation (with and without deformable free surface) show a similar predicted velocity.

In the first section a stationary flow can be seen around the section and in de second an increment of the velocity in the space of between the ships. Between y' 1.38 and 1.46 (Figs 17 to 19) both computations (with a without free surface) are in agreement with the predicted velocity flow, being the flow pattern similar for this range of lateral distances. The velocity at the bow decreases to zero near to the section where the velocity increase highly between the space of the ships going to mid of the ship. This velocity remains constant increasing until x/L_{tug} of 0.44. As in the previous lateral distance, the velocity in the keel is modified in the interaction side only.

The flow at the stern is slightly affected by the tanker and the velocity decreases while in the free side the velocity increases in the bilge as in the isolated tug case. In the cases where the ships have more separation (Figs. 20 and 21), the distribution of flow velocity distribution around the sections is the same observed in the cases of the tug sailing freely (Fig. 4.5).

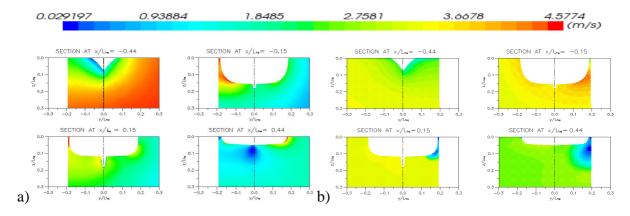


Figure 4.16 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.34: a) Velocity without waves; b) Velocity with wave making.

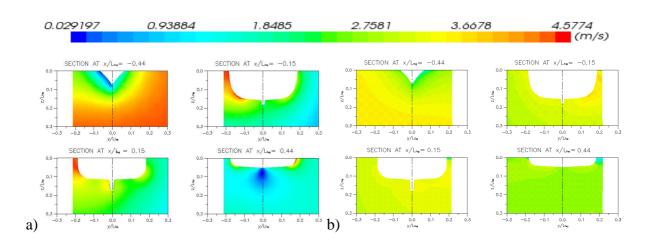


Figure 4.17 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.38: a) Velocity without waves; b) Velocity with wave making.

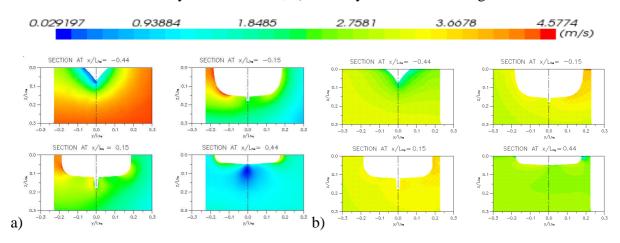


Figure 4.18 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.41: a) Velocity without waves; b) Velocity with wave making.

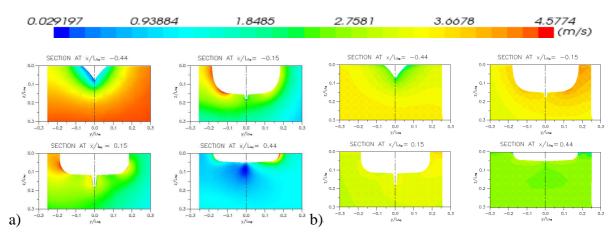


Figure 4.19 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.46: a) Velocity without waves; b) Velocity with wave making.

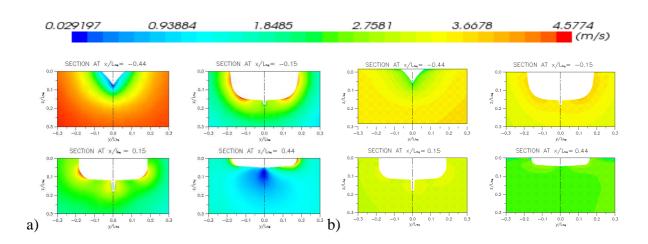


Figure 4.20Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +1.95: a) Velocity without waves; b) Velocity with wave making.

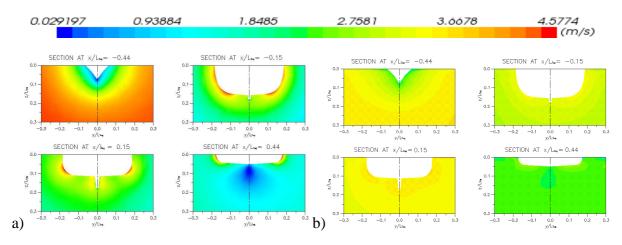
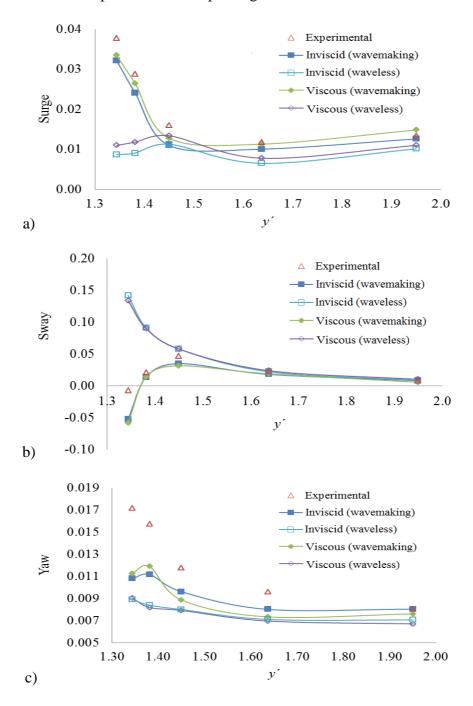


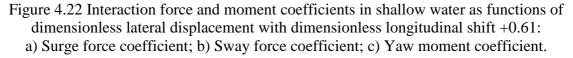
Figure 4.21 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.121 with dimensionless longitudinal shift +0.014 and dimensionless lateral shift +2.26: a) Velocity without waves; b) Velocity with wave making.

4.2.2 Case II (**Fn** of 0.181)

The hydrodynamic interaction coefficients with the lateral distances are plotted in Fig. 4.22. The coefficients have a distribution similar to that in the previous **Fn** of 0.121. However at y' of 1.95 the magnitude of the surge coefficient increases with respect to the lateral distance at y' of 1.64. Thus, the interaction surge coefficients increase the ship resistant in all lateral distances.

The sway coefficient shows a variation in the direction of the lateral clearance where the tug experiences repulsion at small gaps between the ships, however this force is less than in the previous case changing to suction force when the tug vessel increases the lateral distance, As in the other case the yaw moment coefficients have important variations, when the ships are widely separated the coefficients are near to zero. On the other hand, for small gaps, the yaw moment coefficients experience large increments. The numerical coefficients computed with wave making improved accuracy of the interaction forces for small distances although the yaw coefficient (y' = 1.34) was highly underestimated at the same distance when compared with the experiments. The numerical coefficients without wave making gave similar results than in the previous case improving the trend of the curve.





The free surface interactions and their cross sections are illustrated in Figs. 4.23 to 4.25. At the lateral gap of 1.34, the effect of the tanker on the tug is asymmetric. In the gap between the ships, the wave has a short length with higher amplitude than in the free side of the tug. It is noted in the curvature of the bow water cumulated at the interaction side. Near the stern it is noted that the generated wave behind the tug is affected by the tanker, being irregular and asymmetrical.

Fig. 4.24a and 4.24b show the free surface in the lateral distances of 1.46 and 1.64, respectively. Lateral distances of 1.46 and 1.64 have a similar wave pattern than the observed in the previous **Fn**. In the gap between the ships, a valley at mid-section of the tug is generated. The valley generated at the free side (mid-section of the tug) is smaller than in the interaction side. In Fig. 4.25 can be seen the free surface for y' of 1.95. In this lateral distance, the tug is producing a symmetrical wave patterns because the tug is far away from the tanker, and thus the valley in the interaction side has the same peak value than in the free side (z/L_{nug} of -0.0140).

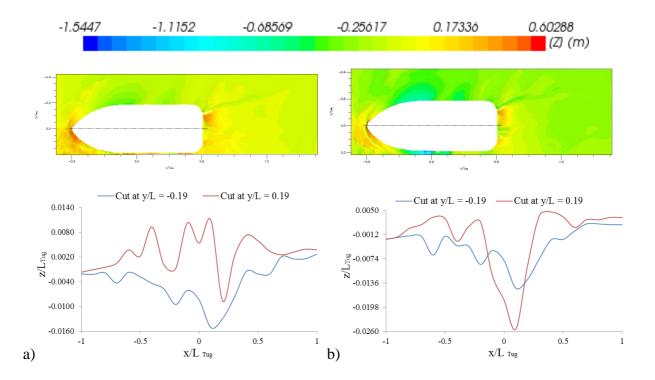


Figure 4.23 Predicted interaction wave patter by the ships for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and wave cut at y/L_{mg} of 0.19 and -0.19: a) Free surface for y' of 1.34; b) Free surface for y' of 1.38.

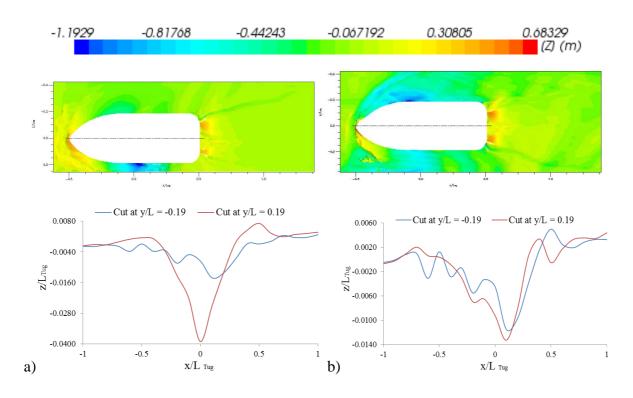


Figure 4.24 Predicted interaction wave patter by the ships for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and wave cut at y/L_{mg} of 0.19 and -0.19: a) Free surface for y' of 1.41; b) Free surface for y' of 1.64.

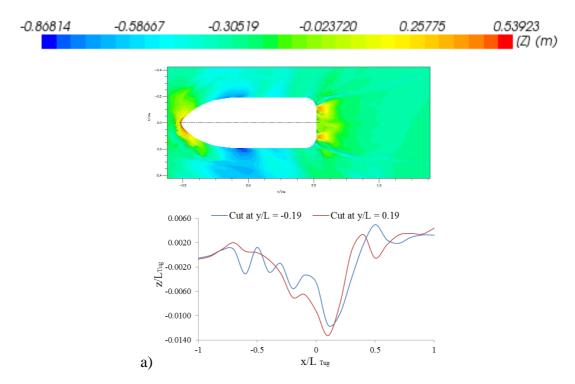


Figure 4.25 Predicted interaction wave patter by the ships for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and wave cut at y/L_{mg} of 0.19 and -0.19: a) Free surface for y' of 1.95.

The distribution of the pressure is shown in Figs. 4.26 to 4.30. The pressure has a similar patter than in the previous **Fn** analysed. At the lateral distance of 1.34 (without deformable free surface) Fig. 4.26a shows a high pressure at the bow, which decrease in de region of curvature of the bow (between x/L_{tug} -0.5 to 0.125). The low pressure in the mid-ship of the tug is producing suction in the tug due to the presence of the tanker. In the positions between x/L_{tug} of 0.125 to 0.5, the pressure increase. However, it is smaller than in the bow pressure. In Fig. 4.26b (with deformable free surface), an increment of the pressure is noted in the interaction side between the ships at the mid-section. Repulsion sway forces accrued at **Fn** of 0.121 where the lateral distance is the same. The high pressure in the bow is generated by the accumulated water. Near the stern, the pressure is close to the initial hydrostatic pressure (when the free surface is non-deformed). In the case of y' between 1.38 and 1.95 (Fig. 27 to 30) the pattern of the distribution of the pressure on the hull is the same in both simulation, being the high pressure at the bow of the tug. It is show in the body view that this distribution of the pressure is higher in the interaction side than the in free side.

The pressure decreases in the bow between x/L_{tug} -0.5 to -0.125. For the stern between x/L_{tug} -0.125 to 0.125 the low pressure is generated on the tug hull, producing suction on the tug hull. The distribution of the pressure increases between x/L_{tug} 0.125 to 0.5 and is constant going to the stern.

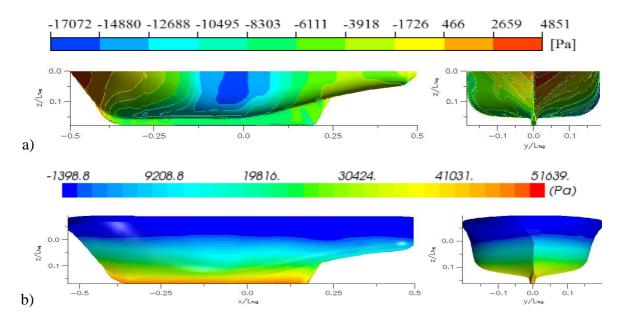


Figure 4.26 Distribution of the pressure on the tug hull surface for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.34: a) Pressure distribution without waves; b) Pressure distribution with wave making.

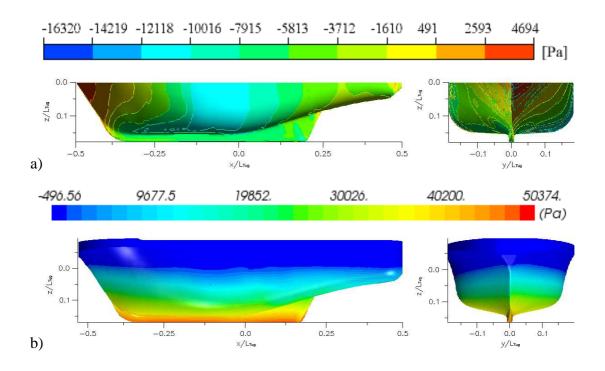


Figure 4.27 Distribution of the pressure on the tug hull surface for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.38: a) Pressure distribution without waves; b) Pressure distribution with wave making.

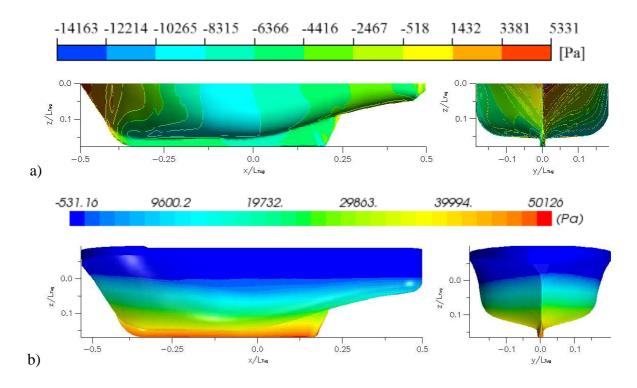


Figure 4.28 Distribution of the pressure on the tug hull surface for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.41: a) Pressure distribution without waves; b) Pressure distribution with wave making.

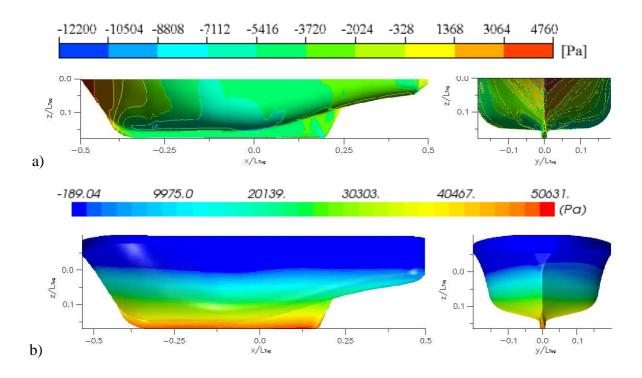


Figure 4.29 Distribution of the pressure on the tug hull surface for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.64: a) Pressure distribution without waves; b) Pressure distribution with wave making.

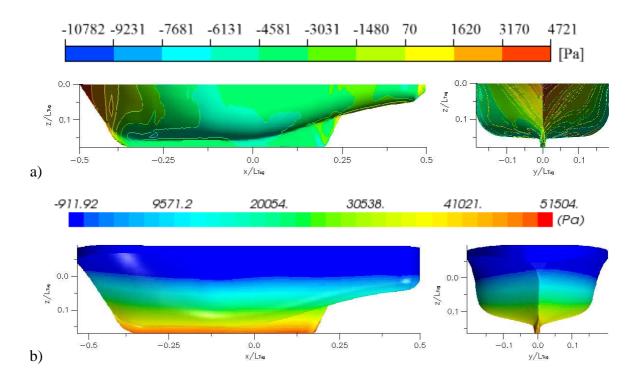


Figure 4.30 Distribution of the pressure on the tug hull surface for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.95: a) Pressure distribution without waves; b) Pressure distribution with wave making.

The flow velocities around the tug are shown in Fig. 4.31 to 4.35, where the velocity of flow represents similar qualitative characteristics as in the previous Froude number calculated. At the lateral gap of 1.34 (Fig 31) with rigid free surface the velocity between the ships in the mid-section increments, while in the other computation is near to zero.

In the remaining lateral distances (Fig 32 to 35), both models predicted similar velocities. In the bow, the flow round the section is stationary whereas going to the mid-section is produced an increment of the velocity between the ships. In stern zone, the flow decreased the velocity at the interaction side while it increased in the free side.

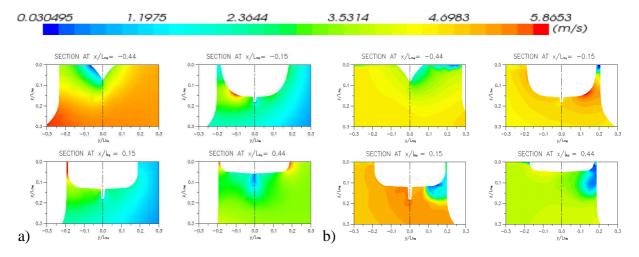


Figure 4.31 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.34: a) Velocity without waves; b) Velocity with wave making.

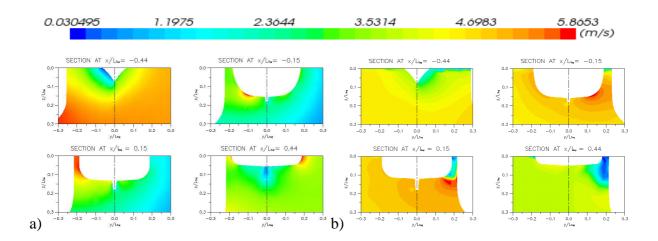


Figure 4.32 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.38: a) Velocity without waves; b) Velocity with wave making.

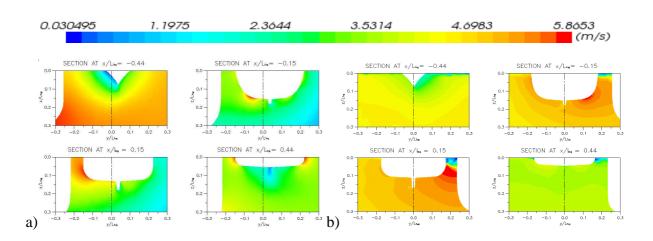


Figure 4.33 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.41: a) Velocity without waves; b) Velocity with wave making.

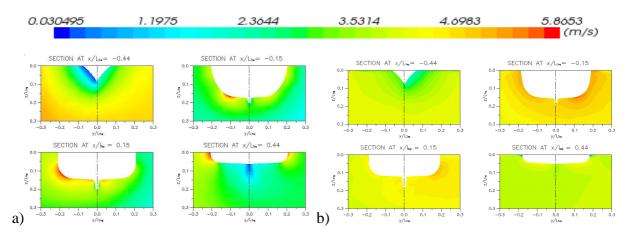


Figure 4.34 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.64: a) Velocity without waves; b) Velocity with wave making.

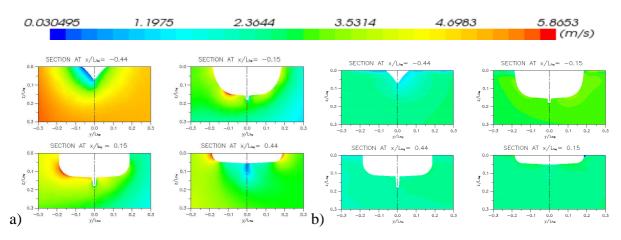


Figure 4.35 Flow velocities distribution around of transversal sections of the tug hull for **Fn** of 0.181 with dimensionless longitudinal shift +0.61 and dimensionless lateral shift +1.95: a) Velocity without waves; b) Velocity with wave making.

The interaction between ships showed that the \mathbf{Fn} of 0.181 is better predicted when wavemaking is used. This is because the deformation on the free surface is higher than in the case \mathbf{Fn} of 0.121, where the mesh selected is fine enough for the VOF method. All movements in the free surface were computed for higher \mathbf{Fn} whereas in the other \mathbf{Fn} the predicted wave patterns were less accurate.

Another problem observed for the low \mathbf{Fn} , is the distribution of the predicted pressure, which were invariable along of the hull. However, the viscous and inviscid fluid computation in general (with and with wave-making) show good agreement between numerical and experimental results.

The behaviour of the tug, when is assisting the tanker, has an associated risk due to their proximity during the manoeuvre. It is observed that the lateral movement of the tug respect to the tanker has a strong influence in the results when the position of the tug along of the tanker is changing or the velocity of the tug increases.

The lateral movement of the tug with respect to the tanker could cause accidents if not proper operations during the navigation are implemented. For example, the suction produced when both the tug and the tanker are sailing very close the risk of collision between them increases significantly.

The negative yaw moment at the bow of the tug is due to accumulated water, which can take over it from its sailing direction. Thus, the risk of hitting the stern and the propeller of the tug in wall side of the tanker increases.

The increment of the hydrodynamic ship resistance is a negative effect. For example, it is necessary an engine with high power to keep the sailing velocity of the tug. In the same line, it was observed that the flow entering on the propeller region is irregular and caused that the propeller lost efficiency.

The information obtained from the analysis is useful to predict the manoeuvre of a tug assisting a merchant ship when the additional forces and moment generated by their interaction is known. The variation of forces and moments can define properly the energy of the engine or the condition in which the ruder could operate. In the current study, the interaction between the ships can be used to avoid accidents. Also it can be useful for the selection of ropes, based on the design loads obtained from the interaction when the tanker is assisted.

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Conclusions

Computational Fluid Dynamics (CFD) models were developed for computing the hydrodynamic interaction forces and moments between two sailing ships models in shallow water. The computations were carried out using viscous and inviscid flow formulations and with deformable or rigid free surface. The computed surge and sway force coefficients and the numerical yaw moment coefficient showed good agreement in general with the experimental results. Both viscous and inviscid flow models predicted almost the same values for the hydrodynamic force coefficients. However, substantial disagreements were found between the numerical coefficients with and without deformable free surface, giving indications of the hydrodynamics phenomena that have influenced the interaction between ships.

The hydrodynamic interaction forces between ships in shallow water at short lateral distance range (between y' 1.34 to 1.41) are dominated by the intensive wave generation and the complicated wave transformation due to the presence of the tanker.

The numerical calculations with free surface showed two main wave effects in the component forces. First, the surge force magnitude in the tug vessel was increased drastically in the smaller lateral distances. Second, a repulsion sway force was generated between the ships at y' = 1.34 for both values of the Froude number.

The interaction yaw moment coefficients showed different effects for different \mathbf{Fn} . At smaller \mathbf{Fn} , the moment increases drastically while at a larger \mathbf{Fn} the moment decreases. This situation is due to the resultant component forces and their location which is defined mainly by the wave positions on the tug. On the other hand, the numerical analysis without free surface at small gaps showed large sway suction coefficients and relatively small surge force and yaw moment on the tug being underestimated with respect to the experimental coefficients.

The interaction forces for large lateral clearances showed that the main source is the velocity potential of the flow, generating additional surge and sway suction forces. The additional yaw moment on the tug was calculated accurately by all proposed models.

Discrepancies were noticed between numerical (with deformable free surface) and experimental values of the hydrodynamics coefficients at smaller lateral distances. This suggested that the meshes in the free surface region were not fine enough and this was most unfavourable for the **Fn** of 0.121.

The longitudinal position did not have strong influence on the results. However, it is mentioned that the tug position at the mid-ship section of the tanker is less favourable. Here the coefficients were not smaller than when the tug was at the bow position while that the velocity was smaller at the first longitudinal position.

The main result of the study concerns the effect of wave making and viscosity on the interaction forces demonstrating predominance of the former. The present work demonstrated the ability of CFD simulation models to quantify the interaction between two vessels in typical harbour manoeuvring.

Also it should be noticed that for a more detailed investigation it would be useful to test a wider range of hull shapes and ship-length ratios for a variety of waterway configurations.

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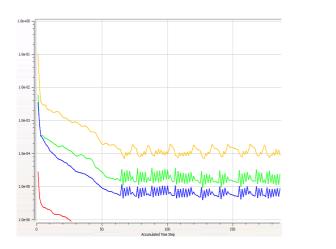
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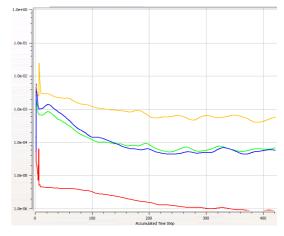
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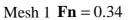
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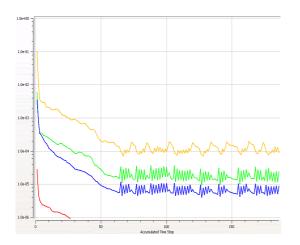


Appendix A: Convergence curve of Series 60 model

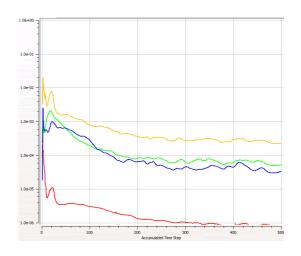




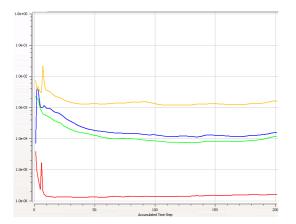


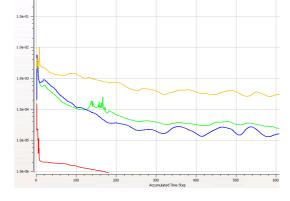








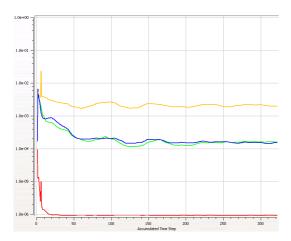




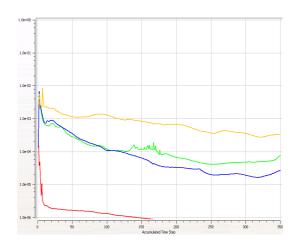
1.0e+0

Mesh 3 Fn = 0.18

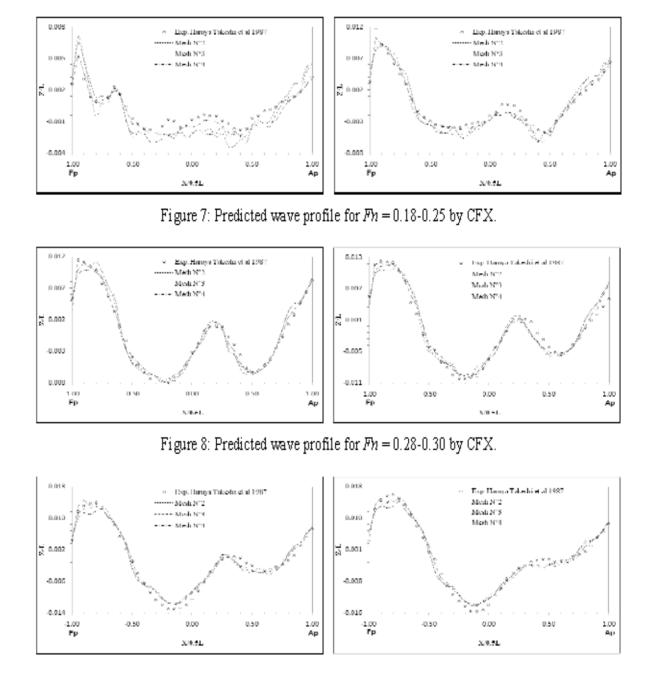
Mesh 3 **Fn**= 0.34



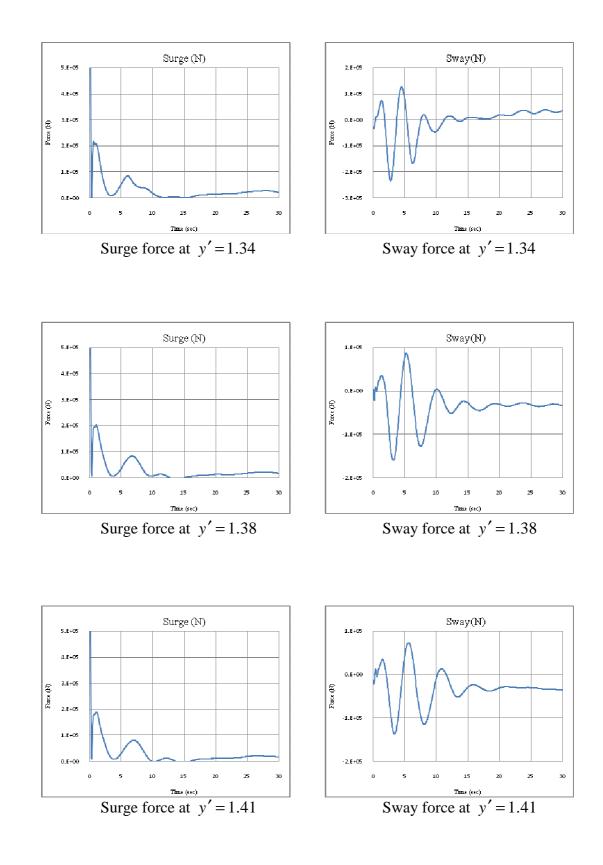
Mesh 4 Fn = 0.18



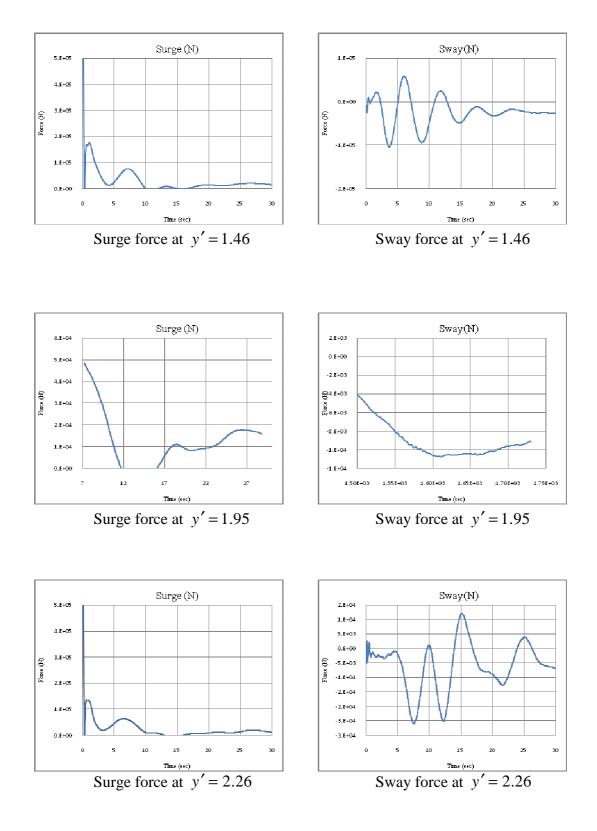
Mesh 4 Fn = 0.34



Appendix B: Predicted wave profile of Series 60 model



Appendix C: Forces monitor at Fn of 0.121



y'	Experimental					
У	X '	Y'	N'			
1.34	0.03972	-0.02000	0.00709			
1.38	0.02321	0.02026	0.00548			
1.41	0.01132	0.04259	0.00180			
1.45	0.00704	0.03381	0.00168			
1.95	0.00418	0.01238	0.00048			
2.26	0.00356	0.01138	0.00054			

Appendix D: Table of coefficients at Fn of 0.121

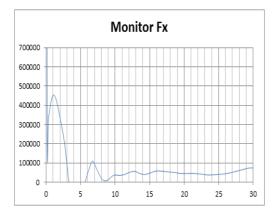
y'	Viscous (vavemaking)			Invicisd (vavemaking)		
	X'	Y'	N'	Χ΄	Y'	N'
1.34	0.02147	-0.05135	0.00363	0.02593	-0.04249	0.00358
1.38	0.01510	0.05077	0.00104	0.00967	0.05713	0.00118
1.41	0.01383	0.05364	0.00071	0.00880	0.04925	0.00078
1.45	0.01149	0.03945	0.00060	0.00807	0.03662	0.00059
1.95	0.00785	0.01296	0.00043	0.00674	0.01073	0.00028
2.26	0.00602	0.01003	0.00028	0.00580	0.01019	0.00012

y'	Viscous (waveless)			Invicisd (waveless)		
	X '	Y'	N'	Χ΄	Y'	N'
1.34	0.01431	0.14013	0.00001	0.01047	0.14943	0.00010
1.38	0.01302	0.08590	0.00099	0.00979	0.09464	0.00143
1.41	0.01297	0.06935	0.00100	0.00902	0.07921	0.00150
1.45	0.01166	0.04702	0.00039	0.00763	0.04857	0.00029
1.95	0.00785	0.00867	0.00008	0.00503	0.00940	-0.00004
2.26	0.00621	0.00649	0.00020	0.00435	0.01024	0.00018

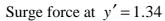
Appendix E: Table of difference between numerical and experimental coefficients at Fn of 0.121

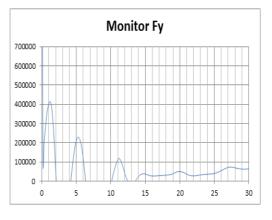
y'	Viscous (wavemaking)			Invicisd (wavemaking)		
У	Χ'	Y'	N'	Χ'	Y'	N'
1.34	-84.971	61.055	95.147	-53.160	52.928	49.490
1.38	-53.699	60.092	81.046	-140.110	64.537	78.434
1.41	18.144	20.598	60.803	-28.625	13.529	56.523
1.45	38.721	14.299	64.286	12.763	7.684	64.592
1.95	46.745	4.489	10.400	37.989	-15.333	40.916
2.26	40.885	-13.443	48.373	38.629	-11.730	78.608

v'	Viscous (waveless)			Invicisd (waveless)		
y	X'	Y'	N'	Χ΄	Y'	N'
1.34	-177.569	114.273	-99.904	-73.643	-847.142	-98.584
1.38	-78.278	76.416	-81.888	-57.802	367.141	-73.915
1.41	12.734	38.586	-44.627	-20.345	85.992	-16.527
1.45	39.619	28.088	-77.019	8.395	43.648	-82.885
1.95	46.770	-42.757	-83.270	20.337	-24.087	-108.202
2.26	42.672	-75.460	-63.462	22.064	-9.997	-67.461

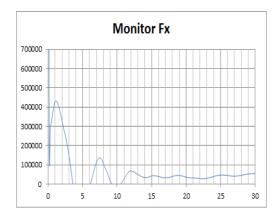


Appendix F: Forces and moment monitor at Fn of 0.181

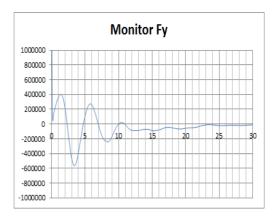


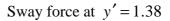


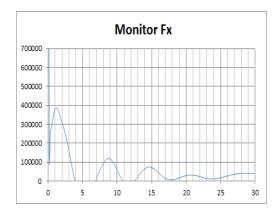
Sway force at y' = 1.34



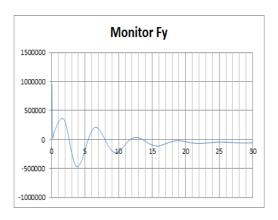
Surge force at y' = 1.38



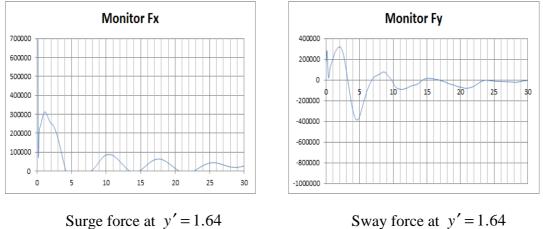




Surge force at y' = 1.41



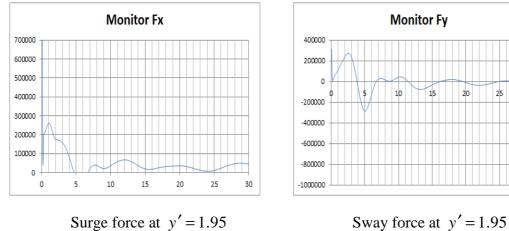
Sway force at y' = 1.41



Sway force at y' = 1.64

25

30



Surge force at y' = 1.95

'	Experimental				
У́	Χ΄	Y'	N'		
1.34	0.0377	-0.0074	0.0172		
1.38	0.0289	0.0214	0.0157		
1.41	0.0160	0.0467	0.0118		
1.95	0.0134	0.0086	0.0081		
2.26	0.0377	-0.0074	0.0172		

Appendix G: Table of coefficient at Fn of 0.181

y'	Viscous (wavemaking)			Invicisd (wavemaking)		
	Χ΄	Y'	N'	Χ΄	Y'	N'
1.34	0.0335	-0.0580	0.0113	0.0322	-0.0520	0.0108
1.38	0.0264	0.0146	0.0119	0.0241	0.0142	0.0112
1.41	0.0128	0.0319	0.0089	0.0110	0.0352	0.0096
1.64	0.0113	0.0193	0.0073	0.0101	0.0184	0.0080
1.95	0.0148	0.0060	0.0076	0.0126	0.0085	0.0080

y'	Viscous (waveless)			Invicisd (waveless)		
	Χ΄	Y'	N'	Χ΄	Y'	N'
1.34	0.0110	0.1342	0.0089	0.0087	0.1420	0.0090
1.38	0.0118	0.0902	0.0084	0.0091	0.0914	0.0082
1.41	0.0134	0.0580	0.0080	0.0113	0.0580	0.0079
1.64	0.0078	0.0242	0.0071	0.0065	0.0225	0.0070
1.95	0.0110	0.0105	0.0071	0.0101	0.0087	0.0067

Appendix H: Table of difference between numerical and experimental coefficients at Fn of 0.181

y'	Viscous (wavemaking)			Invicisd (wavemaking)		
У	Χ'	Y'	N'	Χ'	Y'	N'
1.34	-12.6578	87.2331	-52.3657	-17.0148	85.7650	-58.9655
1.38	-9.0907	-46.3322	-32.2616	-19.5818	-50.9758	-40.8762
1.41	-25.5305	-46.4713	-32.6731	-45.7464	-32.5636	-22.3669
1.64	-4.2003	-19.5382	-30.8329	-17.0985	-25.3904	-19.7178
1.95	9.6917	-43.5400	-5.7018	-6.2800	-2.0288	-0.1066

y'	Viscous (waveless)			Invicisd (waveless)		
	Χ'	Y'	N'	X'	Y'	N'
1.34	-70.8484	105.5144	-93.0328	-76.9344	105.2119	-89.9829
1.38	-59.0719	76.2736	-87.3722	-68.4760	76.5902	-92.1882
1.41	-16.4565	19.5442	-46.9285	-29.5920	19.5232	-48.5338
1.64	-33.7646	4.6721	-35.0260	-44.5737	-2.5312	-37.8019
1.95	-17.9363	17.9508	-13.9485	-24.4276	1.2903	-19.9727